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# AVIATION

*The Oldest American Aeronautical Magazine*



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millions of flight miles in swift, luxurious comfort.

The *Advanced* Wright Cyclones, which will power the new TWA Super Skyliners, are the most powerful single-row, radial engines in the world—capable of producing 1220 h.p. for take-off. These *heavy duty* engines were designed especially for TWA requirements.

When all of the new TWA Douglas Super Skyliners have been placed in service, Transcontinental & Western Air will have a fleet of 90 Cyclone-powered Douglas Skyliners in operation which will cover over a million miles a month, maintaining swift TWA schedules.

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In SOUTH AMERICA . . . on Pan AmericanGrace Airways.

In EUROPE . . . OELAG in Austria, Deutsche Luft Hansa in  
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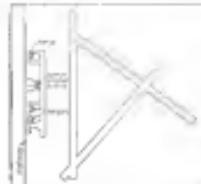
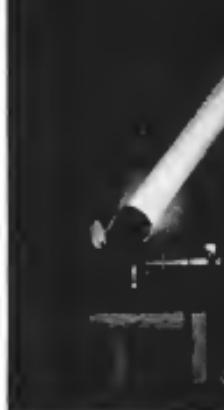
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### SPECIFICATIONS: Beechcraft C17L

Top Speed  
Cruising Speed  
Landing Speed (Gear Down)  
Ceiling  
Service Ceiling

175 mph  
125 mph  
61 mph  
1,000 ft. per min  
12,000 feet

Weight  
Gross Weight  
Empty Weight  
Height

1,025 pounds  
2,050 pounds  
912 feet  
26 hours, 8 minutes  
12,000 feet, 8 minutes

Plus Goodrich Low Pressure Silvertown Tires, Goodrich  
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Committee specifications of the new Beechcraft C17L call for more than its speed and economy performance. These new Beechcrafts are built ruggedly, with a high degree of safety of cockpit seats to the wings, the use of Goodrich Rubber Tubing, Rubber Cements and Goodrich Gasoline Hose as some of the most important features. In addition, "Goodrich" service for Beechcraft owners保障 extraordinary economy, the new C17s actually give better than a dozen miles per gallon of fuel. It's a good idea to get in touch with your Goodrich distributor for more information. Even though this new plane looks as though it is racing planes, even though it has all the leading advantages of other planes—Beechcraft takes the first place of importance in the field of production of equipment with Goodrich Low Pressure Tires. You

only know the type of tire is selected when you buy it, but on other types of tires you might as well buy the good "Goodrich" tires.

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Remember, Goodrich Aviation Equipment creates the entire range of avionics' rubber products. In addition to tires, rubber hoses, rubber cements, there are more than 100 other Goodrich products—everything from dependable airplane equipment to tire products that will increase the safety, efficiency and economy of your aircraft. For the head of the leading pilot, plane and car and air line, "Goodrich" tire products are the best. See your nearest Goodrich dealer or write now to Dept. 60, American Goodrich Corporation, 200 South Dearborn Street, Chicago, Illinois. For complete information



**BEECHCRAFT C17L.** A high performance, economical plane that is a 1938 biplane with up-to-date, all-metal construction. Hand-developed for extra safety, comfort and efficiency.

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**Goodrich Airplane Silvertowns**  
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—Rubber Hoses—Rubberized—Black Aniline Cloth—A Complete Line of Rubber Airplane Accessories.



# The Birdmen's Perch

Chief Aircode and our radio, please. The French know us not yet off with this issue, and we think it's time to make brass. For our fan mail makes us think that the press is pretty popular. So, for Free's sake, don't tell stories on us. Keep on doing all the news and fun and whoppers that make the stories interesting. Address your mail to:

STAN AL WOOD, Director, Aviation Department  
Gulf Aviation Products, Gulf Building, Memphis, Tenn.

IF SHE WERE YOURS



"This Christmas Nancy Bremerton, a girl, will take downvotes to her Christmas tree in any other birthday, except child."

But just a little earlier, a year, Nancy was living happily in New Bedford Hospital (N. Y.), like a real fighting valiant against a streptococcal infection in her left leg, complicated by thrombophlebitis, a blood clot that had formed in one of the leg veins and interfered with normal blood circulation. A rare, recently-discovered serum was used to help her weakened system combat the infection.

When all was no longer than it was then and ready to return to the nurse. There was some in Massachusetts.

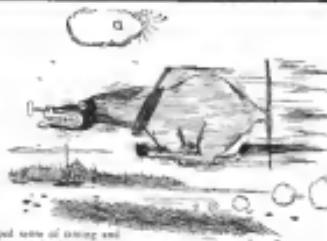
The serum was mailed by plane down Massachusetts to Chicago. From there a plane flew it to Newark. From Newark it was rushed to the desperately ill little girl in the hospital of New Rochelle.

It arrived on time. Nancy quickly responded to treatment and eight days later she left the hospital for home.

"Nancy's is but a single instance out of many. All over the world, day after day, airplanes with their blessing of speed are performing miracles of mercy for countless people."

—W.S.

JUST A LOT OF PICCOLO PLAYERS?  
A great majority of pilots play some musical instrument. Bremerton has such



a finely developed sense of timing and rhythm that it is a joy for them to learn to play a piano or just a harmonica.

—R.E. Conklin

## THIS MONTH'S WHOPPER

James McDonald Honey O'Toole  
Dressed like a hero, see how male  
To start that male he'd try to win,  
But every time just part the same—  
Until one day a Great cause by  
And said, "I'll give that male a try."  
He started the floor, went to the sun  
And soon came back with a small red can  
He opened Sam's mouth and poured some  
in  
And when Sam did was a daggarn' as?  
He turned his tail like a Rummy George,  
And took right off for parts unknown.



Steve Farmer James is the Miracle Man,  
What on the world is in that can?

"To Good Gulf Aviation Gas," says he,  
And held up the can for the farmer to see  
"Well give me a song of that same good  
gas."

"Can't you get to catch that blazed Am  
—E.A. (Bob) Conklin

Gulf Oil Corporation and Gulf  
Refining Company...makers of

**GULF**  
AVIATION  
PRODUCTS



That's a swirl. We're going to get our  
fondness and refine to get us into a trans-  
port and somebody designs one that  
contains a silversmith, a swimming  
pool and a three-chair barber shop.

"Ahh, set your breakfast in our bed  
on Sunday morning."

—H.L.

"All we can meet this summer, the pilot  
will be to buy a new car. The  
purchaser will probably  
suggest that said  
pumper ride on the  
wing. But the young  
fellow had his own  
ideas. 'Not by a demon's  
name,' he yelled. 'I  
won't ride at the  
wing. Why I might fall off!'"

—A.M.



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With higher speeds, higher ceilings and greater ranges, the improved Merlin Bombers open up new possibilities in bombardment operations. Many refinements in aerodynamic detail, and the new G-Service Wright Cyclone Engines, give these enhanced Bombers speeds up to 240 miles per hour, a landing speed of 65 miles per hour, and service ceilings up to 29,000 feet that reveal improvements up and out there. Parallel improvements have been made in facilities for accurate bombing, communications, navigation, and self protection. These improved Merlin Bombers (Series B-10-B, Model 129-W) are available for early delivery.

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BALTIMORE, MARYLAND

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THE GUARDIAN OF AVIATION SAFETY

MORE STRENGTH!  
MORE STRENGTH!  
MORE STRENGTH!  
MORE STRENGTH!



STEP by step, the fast-growing aircraft industry has demanded MORE STRENGTH in each structural part. This greater strength has been vitally needed to accelerate the industry's climb toward its ideal: air transportation, perfectly suited to the needs of those who fly.

Bendix Airplane Wheels and Brakes have always been ready for each new demand, not alone with more strength, but more strength with least added weight.

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AND BRAKES ARE—

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BENDIX PRODUCTS CORPORATION, South Bend, Ind.  
AIRPLANE WHEEL AND BRAKE DIVISION  
(Subsidiary of Bausch Aviation Corporation)

AVIATION for December, 1936

The First

1/2 million  
miles of

**PACIFIC PROGRESS**

By Daniel Sayre  
*Pan American Airways*

AT the great external and jarruline splashes that attended the opening of Pan American Airways' trans-Pacific air mail service last fall and the inauguration of its trans-Pacific passenger service last month, it might seem there was little left to be written on the subject. However, between those two events intervened a period of operations, about which little has ever been published. It was a period of pioneering, a stand-down period, a period in which a thousand technical aspects of one of the country's boldest enterprises had to be perfected in the crucible of half a million miles of transport flying across the world's widest ocean. It is of this period that we offer the following rather fragmentary comments.

For the sake of the record, it is possibly worthwhile recalling the actual achievement in progress. November 22 of last year the China Clipper made what was not only the first round-trip with air mail but also the first complete flight of the route between San Francisco and Manila, the narrow ship Pan American Clipper having penetrated only as far as Guam. December 9 the

Philippine Clipper, which had just arrived on the Pacific Coast, flew nonstop 3,000 miles over the full route. As an entirely tentative working basis, it was planned to then run two round-trips each month for a preliminary period of approximately six months, although the terms of the mail contract and the requirements of the Post Office itself would have been satisfied under a considerably less rigorous arrangement.

The formidable attempt on Dec. 22, the China Clipper, flying a Great Circle course southbound, was forced to come back to Alameda because it had run out of head winds up to 43 miles an hour for almost its entire trip and winds of even greater force were reported ahead of it. For some days the wind conditions continued. Then Jan. 5, while passing out of the headwinds at Alameda for a check flight, the China Clipper struck a submerged obstruction that had been washed into the channel from one of the sandbars forming the breakwater, and had to be towed up to the shop for extensive repairs to its hull. Meanwhile a complete engine change had been started





The Boeing Flying Fortress in  
its final assembly at  
Alameda



A general view of interior showing  
the radio station in  
Munich

on the Philippine Clipper in expectation of the imminent arrival of the third ship, the Hawaii Clipper, from the factory at Baltimore. Now that the China Clipper was laid up anyway, it was decided to change its engine also. The plan however had been made without due regard for vagaries of the climate of Eastern America. One of the severest winters in Maryland history held the Eleven Clipper well-kicked from open water in Chesapeake Bay for long weeks. Not until February 13th did a Clipper, save more the China,

take off for a crossing. And this once were it was forced back by a repetition of the head-wind situation of December 22.

It was a strong gl disappointment that world have broken the hearts of a host of angels. Finally on February 22 the China Clipper got through to Honolulu and completed a restocking on schedule flight over the entire route. The Philippine successfully followed a March 18. Each day made a non-stop round-trip in April.

Then in May the long-awaited and finally-arrived Hawaii Clipper was able to take up its share of the operations. Three round trips were completed in May, four in June, and four in July. Since that period a schedule of weekly departures from Alameda back Wednesday was set with the approval of the Post Office Department and ran through without a hitch until early November when an extra survey flight from March to the Chinese Coast caused a two-day postponement of a scheduled departure.

So went for the trip-by-trip record. Together with the survey flights of the Pan American Clippers and routine non-stop training flights along the California coast, the recall wings flying by the big ships over the Pacific passed the half-million mile mark early in October. After the mid-winter delays there was little more to star the record. There have been two or three instances of minor motor difficulty—use reliable delays in Manila as the early going awaiting engine parts to be lowered in the following Clipper. There have been occasional delays on the Western end of the stage of a day or two due to typhoon or other severe weather conditions—most to the credit then to the deficit side of the operation, since there were proof of the offensiveness of the system's typhoon watch in the Western Pacific.

The stagnation of air mail service had been herald by the country's newspapers as a "final herald of conquest of the Pacific." Those familiar with the difficulties of pioneering any air mail transport system know that it was nothing of the sort. There were so many things that still had to be worked out and perfected that it is almost impossible to even list them all in the space of a magazine article.

The big Clippers themselves were a new departure in aircraft design. Pilots have found that the low center of gravity resulting from the use of the lower hull and sea wings for fuel storage combined with the hydrodynamic qualities of the sea wings make their handle beautifully under all sorts of water surface conditions. Especially when being controlled by the Spruce artificial pilot they have proven very smooth riding in rough sea when we carried, as every one of their flights, within a few hundred pounds of their maximum rated gross load of 22,000 pounds, though the whole operation. The engine, 1,600 h.p. 14 cylinder P.W. Wasp, after the first few months, have given very little trouble and every the Clippers at good cruising speeds over the long California-Honolulu stage at slightly less than 20 per cent of power.

Maintenance schedules, of course, has been a major problem, with only three aircraft available the task of flying weekly schedules over as 8000 mile surface. Under such a regime a ship leaving Alameda on a Wednesday arrives in Manila the following Monday night. The following Thursday morning it begins its return journey arriving in Alameda the following Tuesday morning at approximately noon. From Tuesday noon until the following Wednesday afternoon the big ship will be completely inspected and serviced. Two of the four engines are changed every three trips. The ship must undergo a long test flight on the Thursday preceding each flight and a second one the morning of the day of departure. Even this brief overhauled period is cut down automatically by the full amount of any delay the ship may have experienced on its previous flight, since the character of the route prohibits make-up of lost distance en route. To match the task the maintenance department has paired experienced and skilled crews at each of the Island stations to give each clipper the full benefit of their attention at each mid-Pacific over-night stop. At Manila the Clippers are drawn out of the water and are thoroughly checked in a completely equipped nose-bay. At Alameda itself the facilities have been planned down to the last detail to eliminate any lost motion whatever. A large force of mechanics is departmentalized and specialized in the work so that within six hours after a Clipper has arrived it is in the shop and being worked on by several dozen men,

math going along a familiar and preplanned assignment. Pads at every maintenance stage in the division are the four big working platforms, two at Alameda and two at Manila, that are moved under each wing to speed the work. No more washing ladders these. Each is a complete and highly specialized workshop in itself. Their big top decks have work berths containing specialized tools, special recesses to fit around propellers and engine, electric and compressed air outlets, pumps for oil drainage, and so on. Beneath are air compressors, tanks, compartments for cleaning equipment, even and finally a kitchen sink for the use of the crew who specialize in keeping the Clipper cabin in spick and span shape. Through the year an ample maintenance

(Turn to page 67)



A typical radio room in the radio station



Pan American Airways Clipper in flight



Where radio equipment passes the ultimate

Caliber six to  
poured on a Pacific  
island



expansion, their use is necessarily limited to situations that are to go into large production, and in general new production methods stress the importance of designing in the selected elements at as many different points of a given machine as possible. Mr. Hough- ton had it down as almost a universal rule that "any part can be machined from raw material by limited production, it can hardly be converted to a design for large production."

The Von Hake paper drilled upon the importance of careful planning of fixture, layout and tooling and upon the importance especially in a medium size factory of getting machines flexible enough to adapt themselves to many different jobs and in a wide range of working materials. Punch presses, drop hammers, and hydraulic presses were the most important of the type machines used. With the fixtures so equipped, the author recommended that tooling be kept as closely as possible seeded, with all jig, fixture, and tool designs and storage handled through a central agency.

#### Skips with Queen Mary since peace

In a section devoted to designs in the heavier sizes, and not aiming to run production, Mr. E. K. Ball, chief engineer of Douglas, presented a master review on "Problems in Design and Construction of Large Aircraft," written by R. J. Minshall, chief engineer, Fred. D. Lasham, and Mr. Ball, himself. Building experience was reviewed from the 1925 Memphis, three of the intermediate commercial monoplanes down to date. The authors declared that their unswerving allegiance to the biplane wing construction was good, but they agreed that the monoplane had much inferior type. As the last 164 larger airplanes which would still fit the same structural practices that they had approved over the past seven years.

Though they found large surfaces probably difficult in some respects, mostly in their low frequencies of vibration and consequent likelihood to resonance with forced periods, they found some merit in the monoplane. In particular they believed large gaps greatly to reduce trouble with oscillation, as in highly tapered wings. Some of the arrangements present doubted that and demanded proof not as yet forthcoming.

The authors opposed control of large airplanes through mechanical controlling gear totally separate from the controls themselves and recommended either the use of servo-type. Boeing had tried these but very successful, and likely to be acceptable for all large airplanes now in sight. Several mechanisms to increase braking power were also proposed and proposed air had been used, giving plenty of power but no movement that big. On the Boeing four-engined bomber the amount against braking power had been of course rather than of braking, as the wheels could easily be slid with very little effort for the takeoff.

Structural design of the largest airplane must be based mainly on rigidity, on the other hand, Boeing had found that a more perfect full stress modulus in the skin plates was used by other manufacturers in their wing skins, as employed tubular elements as far as possible, it kept the thickness of the outer skin of a wing at least half that of the over-gaged material underneath, underlying, as the nature of good torsional strength is to reduce wrinkles in the skin under tension within its span. He believed in balancing both aerodynamically and dynamically by elements, so that each part of the aircraft would be completely balanced within itself. His methods were quoted as favoring high consistency of stability and dynamic balancing, finding neither one sufficient by itself. Mr. Ball found at least that balancing was not sufficient, as it could be valid with respect to only one particular set of axes.

Edward Burton of Douglas detailed the severity of wind shear forces for example, as the weight went well beyond 30,000 lb. He recommended accurate aerodynamic balancing to give light control by the ordinary direct brakes as in that point, or even higher. He said the customary cold air-gage stabilizers to produce power plants of higher unit output for these very large airplanes, and he disagreed by implication with his associate, Mr. Houghton, in suggesting cost savings by using a single engine where Mr. Houghton had wanted to do away with one and use light alloy aircraft.

As an example of the dangers of natural frequency with increasing size and consequently increasing gravity as a design problem, figures were given for a model of airplanes. For a fa, for example, the natural frequency is 3.008 Hz if the airplane is 1530 cycles of vibration per second. In a machine of that size that weight it had failed to 650 and the increased liability to resonance, other things being equal, would be obvious.

In conclusion, the authors observed that their type should have reduced to "larger" than to "large" aircraft, for they believe that the large airplane of today would be "but a stepping stone to airplanes of immense proportions." Passageway of the present type would disappear and all equipment and load would be housed within the engine section of the wing. They also believed that the flying boat has no limitations at all to size at the present time. The

projection onto the future led to a lively discussion, and in some propellers of which the most romantic estimated that the Queen Mary would be just about the right size for a wing-tipped float on the future flying boat. Though one or two spoken thought there might be a limit to economical size, none of them was bold enough to attempt to set it.

The writing paper had advocated aerodynamic control, where a mechanically parallel, constant moment control force. That showed an argument from the advocates of dynamic balancing as a better supervisor. Dr. Klein of the California Institute of Technology thought reversibility could not be applied only at the control arm, as the aileron or elevator could cause oscillations within its span. He believed in balancing both aerodynamically and dynamically by elements, so that each part of the aircraft would be completely balanced within itself. His methods were quoted as favoring high consistency of stability and dynamic balancing, finding neither one sufficient by itself. Mr. Ball found at least that balancing was not sufficient, as it could be valid with respect to only one particular set of axes.

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#### Box car or Doge's palace

Arthur E. Raymond, chief engineer of Douglas, digressed from his printed paper long enough to express agreement with Mr. Ball and his associates at least on the possibility of new designs any upper limit to the useful use of flying boats, but Mr. Raymond's paper was concerned more with passenger comfort than with aerial size or

(To be continued)

# On Top

**Airline operation up to 15,000 ft. now involves little difficulty. Beyond that point mechanical and physiological troubles multiply rapidly, but remarkable economies are so attractive that many air lines are considering means of pushing operating ceilings up toward the 30,000 ft. level. Here is the record of TWA's high altitude research to date, by the man who has been doing the work.**

#### By D. W. Tomlinson

Assistant to the President  
Transcontinental & Western Air Inc.

WHEN TWA decided to undertake an investigation of high altitude flying, two planes were available—a single engine Northrop Gamma and "200," the biennial DC-1, last Douglas transport to be built. The performance of either plane at altitude is approximately the same, but obviously not the operating time, but the value for the low plane is far less for the DC-1. The value of results obtained, however, lies in the completeness and accuracy of data obtained during the flights. It is impossible for a pilot to fly an airplane manually with the greatest possible accuracy in test work, and, of the same time, there are many measurements and record data. The DC-1, however, was built to be used in a crew cabin to do the job of observing and recording data. Also in this plane is equipped with an automatic pilot, the pilot is free to handle the engine controls, a job which in a large airplane on test flights of the nature contemplated, demands constant attention.

Says it was desired that the airplane be capable of operating with

normal gross load (16,000 lb.) at 25,000 ft. or above (approximately due to altitude ceiling for the DC-1 with Cyclone 3-8 engine); a higher degree of supercharging was necessary, and two Wright Cyclone F-18 engines equipped with two-speed blowers, having a critical altitude in high-speed blowers of 16,000 ft., were installed. These engines not only gave the required ceiling for the plane, but also at 20,000 ft., had a maximum power rating of 1000 hp (5000 ft.) at sea level to 30,000 ft. This was essential in order that data could be obtained to show the actual increase in velocity of the airplane with increase in altitude with power maintained constant, for comparison with the theoretical increase upon which predictions have been made

concerning operation in the stratosphere.

In order to obtain low propeller and engine efficiency, Hamilton Standard controllables with constant speed governors were installed. (See AVIATION September, 1936.) The governors may be set to hold the engine at any desired rpm and, thereby prevent the engine to operate at maximum efficiency for any desired condition from full to crating power. The constant weight loads which limit the pitch of the propeller should be increased to increase the pitch range to 20 deg instead of the normal 10 deg of the standard two-position type. The high pitch of the standard two-position propellers is 32 deg. In order to hold the engine at a certain rpm, reducing loads at 20,000 ft., and above, the high pitch limit of the constant speed propeller



**TWA**



such storms have their roots in the extremely turbulent clouds forming the general storm top at about 22,000 ft. The plane was flown directly through one such thunderhead at 26,000 ft. The temperature on the ground at Kansas City at the time was 100 deg. F. The temperature within the aerial head was 5 deg. F. The cloud moved at the rate of 50 miles per hour, and water particles which froze out formed ice plates which clung to the front of the plane. Formed a conical ring of ice about 4 in. in dia. as diameter, but so agreeable seemed the ice formed. The air in the aerial head was apparently rising at 300 ft. per minute according to the rate of climb indicator. No descending current was encountered. The aerial head was encased in a semi-transparent covering (water freezing outside) of about 1/8 in. thick. The rate the distance through the cloud was about 16 miles. The upper portion of the aerial head was toward the south. The outer surface of the overhanging portion of the cloud was estimated to be above 35,000 ft. and the ultimate top of the storm probably reached in 35,000 ft. or into the base of the stratosphere. Observations indicated that the top of the turbulent clouds containing the storm descended toward the advancing wind and drift low over at about 35,000 ft. at which altitude a plane could have just crossed the north-south line staying "on top."

The wind over Indiana, Illinois and Ohio held steady at west 75 mph as the storm was approached. From west

On the flight of the DC-2 from Kansas City to Newark, in August, crossing for the most part at 27,000 ft. we had excellent view at the tropical hurricane which had moved up the east coast and was then centered roughly over Virginia and the Carolinas. The upper桑桑 of the storm was situated approximately 100 miles south of New York State. The top of the coney was shaped like a squat umbrella. The top at the center was estimated as 35,000 ft. This and its rotation at the outer fringe, this mass of cirrus clouds descended until it passed the turbulent lower layers at about 30,000 ft. It was a voluminous looking mass of angry air which appeared black, purple and almost green in places. This coney lay about 150 miles or more east of the center of the plane at 35,000 ft. The outer edge of the coney, lower in air spread out in all directions tapering downward from 35,000 ft. at the center to about 15,000 ft. over central Pennsylvania. This semicircular layer is broken about 100 miles out from the center and disappears completely within another 10 to 20 miles. The DC-2 began descent through this semicircular layer south of Philadelphia, Lancaster layers and scattered clouds were found behind it. This coney disappeared with increasing altitude as the remainder of the way to Newark.

The wind over Indiana, Illinois and Ohio held steady at west 75 mph as the storm was approached. From west

ern Pennsylvania, where the plane entered the influence of the mass the wind piled up to about 80 mph and became more northerly as the plane progressed eastward.

#### Where ice was found

As a result of these observations it is considered that severe weather conditions involving considerable turbulence and heavy icing may be found all the way up to the base of the stratosphere. In general, however, a plane arriving at 35,000 ft. can remain on top by making inconsequential descents. A flight of 1,500 miles or more (full way across the country) descending 30 to 50 miles to either side of the west coast does not involve a serious loss of altitude.

With the start of the DC-2 the high altitude program was continued using the Northrop Gamma equipped with a special Wright Cyclone engine and an exhaust turbo supercharger. Approximately twenty hours of flight testing at altitudes up to 38,000 ft. have been carried out using this airplane. The oxygen system with the Gamma paralleled that of the DC-2.

The Gamma has a crew of one, a pilot and an observer. The observer is in a cockpit just forward of the pilot's cockpit. In the observer's cockpit there is a complete set of engine flight instruments so arranged that data is recorded by photographing the instruments board using a Leica camera. The observer takes pictures at predetermined altitudes and later the film is developed and the altitude data recorded from the film. This has been found to be the only practical way of obtaining a resolution reaching at the 2% increments.

#### Wind at stratosphere base

In order to accurately determine surface speeds and the force and direction of the wind at high levels the plane has been flying a triangular course from Kansas City to Rockford, Columbia and return. This triangle is 336 miles. It is marked by radio beacon which permits the plane to be flown on radials or on top. In this way the calculated speed of the airplane at altitudes up to 35,000 feet has been checked within two miles per hour of the true wind speed from the indicated airspeed. Wind velocities up to 40 mph have been observed under favorable conditions. It is planned to fly this triangle with the wind to certain wind conditions exist at low levels in order to determine wind velocities at the base of the stratosphere during these conditions.



## Practical Uses of PROPELLER BRAKES

By Raymond B. Quick\*  
*Editor of Air Commerce*

ice piled up on a long range flight, the propeller brake provides weight assistance for at least one extra person.

The brakes also have been very beneficial for reduction of landing speeds. On recent tests with a large four-engine flying boat it was shown that a 10% increase in landing speed is obtained with one or more dead engines. Secondly, it may be the means of saving a ship from serious structural damage in the event of certain types of engine or propeller failures.

At first propeller brakes attracted manufacturers and operators as a means of increasing performance at high or average altitudes on twin engine aircraft after the failure of one engine. This was a natural development, however, that on some two-engine aircraft an increased altitude over that with full engine power at high pitch will be obtained. In fact, when brakes were used on either engine or one of the large transports there was no gain in loss of altitude, but a marked decrease in vibration.

There have been, however, several developments in the use of propeller brakes which are an appreciable increase in speed or load flight when the propeller of a doubled engine is restrained from rotating in high pitch. This was demonstrated in tests made with the flying engine switched off and the propeller set at high pitch. It was found in specific conditions, on one large four-engine flying boat, that locking out propeller rotation permitted a marked increase in speed or load when the brakes were applied. In addition the tail surface vibration from disturbed airflow with spinning propellers was completely lacking.

#### Controllability increased

Another important detail is the increase of controllability, longitudinal and directional, when the normal propeller is stopped. Less rudder is necessary to maintain directional flight when the dead engine is braked and, as there is no disturbed air behind the wing over the wing, the lift of the wing area behind the propeller disk seen is not impaired. Thus little

or no adverse trimming is necessary to maintain lateral flight on an even keel. Therefore, when a landing is made at the destination devoid of a stopped engine, there will be an increased amount of controllability. This is especially true in landing aircraft using high as most of the big transports are almost directly the engines.

In addition to these factors, there is an increase in safety resulting from the use of propeller brakes. Even with a most proficient captain for the protection of our engine manufacturers, we must be prepared for the unexpected. The safety of every airplane depends so much on man's factors, that it is serious to not consider prop. aids when we consider valuable equipment.

In flying boats, particularly, it is almost impossible to consider landing on the open sea, especially at night, for the purpose of making repairs in a disabled engine and it is imperative that the aircraft be able to continue flight after losing one or even two engines. Unless the disabled engine is easily stopped, disastrous whirlwind set up by the spinning propeller or the unbalanced condition of the engine will be at least partially to blame.

Propeller brakes have been considered favorably by all aircraft manufacturers, but the constantly increasing engines which have lesser more or less response, has caused serious objection to anything that might separate the weight range. The value of using propeller brakes at present is a matter to be decided on by the captain operator and the career of his aircraft.

\*The development of propeller brakes which is described in this article has been the result of the Air Commerce Division of the National Research Council of Canada. The author wishes to thank Mr. T. H. Black, Director of Air Commerce, for valuable assistance.



George H. Jones (left), Ray B. Quick (right) Standard Steagor, the author. Photo by Morris

# How FAST?

## II.

### Ultimate Maximum Speed at Altitude

The second of a series of articles discussing the upper limits of airplane performance. Future articles will cover range and ceiling

By  
**Dr. Norton B. Moore**  
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 and  
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the moving airplane flying at sea level  
The above equation then becomes

$$V = 125 \left( \frac{R}{J} \right)^{1/2}$$

where  $R$  is the ratio between the density at the altitude considered and that at sea level and where  $R$  is the percentage by which the maximum speed is reduced due to the increasing effect of induced drag. In the first paper, the quantity  $\left( \frac{R}{J} \right)^{1/2}$  was called the "core" of speed.

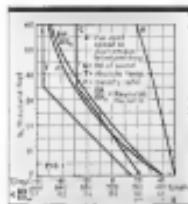
It will be seen from the equation for high speed at altitude that, neglecting the increasing effect of induced drag, the "core" of speed decreases in proportion to the density ratio.

#### Effects of increasing altitude

In order to study the problem of high speed performance at altitude, we must

consider three important factors. First, the air is less dense at increased altitudes; second, the increase in parasite drag sufficient due to the decreased Reynolds' Number; and third, the increased effect of compressibility. In Fig. 1, the physical variations of the quantities which control these three items are plotted against altitude. The density ratio was taken from the 60th issue of aeronautical tables and the variation of performance with altitude from N.A.C.A. Technical Report No. 216, 3, second item which must be considered is the changing scale effect with altitude which can be represented by the Reynolds' Number. The "Reynolds" Number is the ratio  $\left( \frac{V L}{V} \right)$  where  $V$

is the velocity of flight of the airplane.



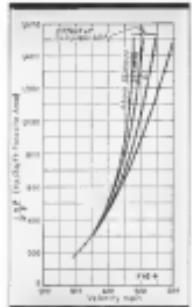
#### AVIATION October 1936

$L$  is some characteristic length of the airplane such as the wing chord, and  $V$  is the so-called kinematic viscosity, i.e., the ratio  $\mu$  to the absolute viscosity.

Assuming constant speed, the variation of the Reynolds' Number with altitude for a given airplane will depend only upon the variation in kinematic viscosity,  $V$ . The results of this calculation are shown by the uppermost curve in Fig. 1 as a ratio  $R/V$  to 100,000, the Reynolds' Number at sea level. The third factor, influencing speed at altitude, is the changing velocity of sound. The velocity of sound is proportional to the square root of absolute temperature; the variation of which is shown in Fig. 1 by the curve  $T$  and is given in degrees Fahrenheit,

thus the approximation being very close.

Dr. Th. von Kármán, in the first issue of the "Journal of Aeronautical Sciences," gives a relationship between skin friction drag coefficient and the Reynolds' Number for turbulent flow, and the portion of this relationship which concerns us is reproduced in Fig.



at sea level showing the added "core" incurred because of skin friction. While we are considering performance at altitude, we must use what we might call the altitude "core" of speed

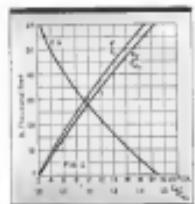
$$\left( \frac{C_f}{J} \right) \text{ instead of } \left( \frac{C_f}{J} \right)_0$$

the velocity of sound decreases with altitude as shown in Fig. 1. It is evident that at sea level  $C_f$  is about 0.0015, but at 10,000 ft,  $C_f$  is about 0.0008.

Calculating the effect of the variation of the velocity of sound, together with the variation of the Rayleigh Number with altitude, the ratio at sea level of the core of speed at altitude to that at sea level,  $C_f$ , is obtained and is plotted in Fig. 2.

Not all of the parasite drag of an airplane, however, is made up of skin friction drag. Part of it is a result of so-called form drag which depends upon the form and size of the body and the angle of attack. It is evident that as the form drag does not vary appreciably with altitude, the effect of induced drag, however, is made up of skin friction drag. On this assumption, the effect of induced drag at altitude can be calculated from the core of speed at sea level and the variation of the core of speed with altitude which is given in Fig. 2.

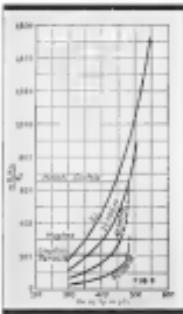
Induced drag. The relation between the ratio of Reynolds' Number at altitude to that at sea level has been given. Its effect on  $V$  must now be calculated. If we assume an airplane with a wing chord of 4 ft



#### Second velocity and speed

In our earlier paper, curves were presented giving the relationship between speed and the "core" of speed,  $\left( \frac{C_f}{J} \right)$ .

In the first section of the paper, an equation was given which satisfied a factor,  $R$ , which represented the percentage to which the velocity would decrease due to induced drag. It is possible to show that as the very high altitude induced drag would play an extremely important part. Actually in the case of racing airplanes this decrease is not as great as might be expected. It is possible to calculate numerically the percentage of decrease in speed due to induced drag for all airplanes since this factor is different for each airplane considered. There-



fore, reasonable values were picked based on what the author considered to be an ultimate design, and a variation of this factor is noted in Fig. 5.

#### Total "Cost" of speed at altitude

Interpreting the factors mentioned above except that of compressibility, Equation 2 may be written as follows:

$$F = M \cdot \frac{1}{2} \rho \left( \frac{V}{C} \right)^2 \left( \frac{L}{C} \right) + R$$

It is the value of  $\left( \frac{V}{C} \right)^2$  (three

hp/sq ft of equivalent parasite area at sea level) plotted against speed for the aircraft in Fig. 5. This curve represents the speed in progress at which the effect of the compressibility. Fig. 5 is obtained. It becomes immediately apparent from this figure that as higher and higher altitudes are obtained, the "cost" of that speed becomes considerably reduced, but the maximum obtainable speed by any aircraft will also be reduced. In this figure are noted the three racing aircraft referred to in our last paper, so that the advantage which might be gained by going to altitude can be determined.

It is of interest to plot the percent age increase in maximum speed attainable for each of these three aircraft. This is done in Fig. 6, and the result is extremely interesting. In this figure, the solid lines are calculated values and the dotted lines represent a reasonable extrapolation of the solid curves to higher altitudes. Assuming constant thrust, however, available from the maximum possible thrust to which these three racing planes could

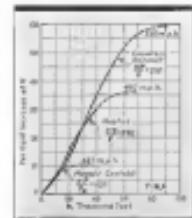
be extended from the endpoint of speed. Along each of the curves are noted the values of  $\left( \frac{V}{C} \right)^2$  and also

the speed attained at the peak of these curves. The most interesting feature on these figures is that all three of these aircraft, at a distance of 1000 ft, would have maximum possible speeds within 20 mph of each other, assuming, of course, that the thrust horsepower might be maintained independently of altitude and without increase in the equivalent parasite area. This, at least, is in a very doubtful position at the very high altitudes, but this is the most interesting that by means of a racing driver's ingenuity, it might be obtained at reasonably high altitudes. The maximum speeds for the three aircrafts, of course, are obtained at very different altitudes.

It might be suggested from the maximum speeds the author has in mind that some range of speed which the airplane could exceed. However, in the case of the Macchi-Cantieri, a very great amount of power was obtained, but at the same time, there was a considerable loss of aerodynamic efficiency. In the case of the Hispano, the author has extremely little except for the maximum obtainable speed, which, as shown in our first paper, is not greatest when ultimate speeds are desired. The Coanda-Bailey was extremely clean aerodynamically but had a comparatively small amount of power available per unit of frontal area. Thus, it is not clear whether any one could come up with desirable features of either, a definite increase of the same speed could be attained.

It is apparent from the last figure that as the record speed in altitude gains, the altitude at which it will be attained will be increased, thus, leading to the obvious conclusion that the ultimate altitude at which the aircraft can attain a maximum speed will be attained at sea level in accordance with our first paper. This does not mean,

that the power could not be obtained independently of altitude, and at present, there is shown no method of calculating. However, one, however, the "cost" which must be paid is at a sort of tax level of compressibility. The equation above represents the result of the maximum possible aerodynamic economy. That same equation will be used to advantage in a more practical calculation in the next paper which deals with the maximum range and endurance, in which economy is the key note.



# Thin-Walled STRUCTURES

By F. R. Shanley

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The author of last month's article, "Pounds or Pounds per Square Inch" continues the interpretation of modern forms of construction on a weight-strength basis and gives us his view on future trends.

MODERN designers are concerned with two general types of hollow structures, namely, the "thin-walled" (Type I) and "solid" (Type II). Considering different shell types of structure of the same diameter and cross-sectional shape, it was found that the weight/strength factor (strength being held constant) for the two latter types of factors were given by the following equations:

$$\frac{W}{S} = C \left( \frac{w}{T} \right)^2 \quad (1)$$

$$\frac{W}{S} = C \left( \frac{w}{T} \right)^2 \quad (2)$$

In these equations  $w$  is the unit weight or density of the material,  $T$  is the thickness (inches) at yield,  $C$  is the safety factor, and  $S$  is the ultimate stress. In the case of the thin-walled structures,  $T$  is a measure of the stiffness of the structure, and  $C$  is a constant used to denote proportionality. In deriving equations (1) and

other useful equations was found:

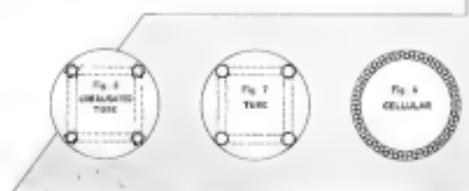
$$\left( \frac{W}{S} \right)_{II} = C \left( \frac{w}{T} \right)^2 \quad (3)$$

where  $T$  is the shell thickness.

With these three formulas we are prepared to evaluate different types of construction in a general way and can possibly predict the future trend in structural design. (The weight/strength factor should be as small as possible for the most efficient design.)

#### The monocoque

Let us assume that the designer has decided on the external dimensions of the structure and wishes to use a fuselage construction of the shell type (see Fig. 1). If a true monocoque (egg-shell type) is chosen, the question of what material is best on a weight-strength basis is readily determined by direct application of equations (1) and



(2). For instance, it was shown in the previous article that a solid shell would weigh almost four times as much as a wood shell of equal strength, assuming that a true stability failure occurred as was possible in a frame type of structure. This would indicate the absence of unreinforced outer shell construction among such structures. Even though a solid shell might offer many practical advantages over a wood shell, the increased structural weight would be too great a penalty to overcome.

The problem would be less however, in aluminum alloy, the weight of the solid shell as this may be about 2.5 times as great as that of the wood shell.

The problem would be further reduced by a complete weight analysis covering the weight of forming rings, frames, and such. True, so far as it is at present to find among our aluminum alloy structures, a comparable number approaches the true monocoque type.

We can take up any of the other commonly used types of construction on this basis. It will be necessary to expand our theory to cover what might be called the *Joint and plate-type* construction. Here again, it will be recognized that *case 1* designer has made up his mind to use a certain material, the problem becomes one of deciding what type of construction is most efficient as a weight-strength basis with due regard for the many practical problems which cannot be avoided. The first principle involved has been clearly stated by Dr. Tiedemann in his excellent paper, "Aerodynamics of Thin-Walled Aircraft," which he gave at the "Aeronautical Meeting, Toulouse," in 1936, when he says: "The (impermeable) joints are to receive the material as far as is possible from the axis of the structure as far as structural elements and at the same time holding it firmly together."

The second principle might well be a rule that the *effective thickness* of the shell should be made great enough so that the failing strength of the material will be approached rather than exceeded. In other words, we can theoretically increase the "stability" strength of a structure indefinitely by spreading out or expanding the material so as to decrease the shell loss, or would be less liable to concern that process beyond the point where a "stress" failure would occur. This process of thickening the shell by expansion of the original material simply amounts to reducing the effective value of  $\pi$  in equation (2) more rapidly than we reduce the value of  $VE$ , thus increasing the weight-strength factor. (In this method of analysis we have all properties of

constant cross-sectional area, therefore  $\pi L$ , which is in terms of a base per unit area, will decrease with increasing  $VE$ .) It will be seen that this process does no good to the shell as a whole, but it does not do a bad job, indeed, proportionately, in equation (2).

It can now be seen that all forms of solid shell construction represent the designer's efforts to decrease the weight strength factor for "stability" failure by expanding the shell material as much as possible, thereby increasing the effective thickness of the shell and reducing the effective area and increasing the overall weight. It is often an effort, however, in the design where the weight-strength factor for "stability" failure becomes about as good as that for "stress" failure. Obviously, for materials like these, and which have a good yield strength on a stress basis, but which are inherently "brittle," the shell-thickening process must be extensively used in order to realize the high strength of the material.

#### Evolution of thin-wall structures.

We can now estimate the various forms of construction in the light of the above outlined analysis. Thus we can distinguish between two categories of shells, which, unfortunately, always seem to appear together in the literature (example, biplane). Without going into the mathematical derivation it may be shown by the methods employed in the previous article that the weight-strength factors for shells used as columns are approximately as follows, for a constant strength basis:

$$\begin{aligned} \text{Local stability failure (downdown)}: & \frac{1}{\pi R^2 V^2} \quad (2) \\ \text{Local stability failure (up-down)}: & \frac{1}{\pi R^2 V^2} \quad (4) \end{aligned}$$

$$\begin{aligned} \text{Buckling stability failure (L and D)}: & \frac{1}{\pi R^2 V^2} \quad (5) \\ \text{Buckling stability failure (L and U)}: & \frac{1}{\pi R^2 V^2} \quad (6) \end{aligned}$$

A complete interpretation of the above factors is beyond the scope of this article. The engineer will recognize, however, that the modulus  $E$  plays a very important role in the determination of stability factors and we shall discuss this factor later. For instance, equation (6) indicates that the skin in carrying compression loads and the tendency to expand as a result of adding the columns. Even the formers used in modern solid-shell aircraft structures reduce the tendency to expand. Thus, the factor  $L/D$  which can also be considered as a ratio of the effective or carrying load. The expert in these methods of analysis may find it possible to increase the factor by means of the weight-strength equations by regarding such formers as a method of improving the "stability" equation with terms of the "stress" equation. (It may be significant that the term  $V^2$  plays a prominent part in all good problems and also in the weight-strength equation for stability failure.)

There seems to be a sort of cycle in the development of airplane structures, as we report them in the light of this new method of analysis. For instance,

Figures 1 to 8 illustrate the progression from the plain shell (monocoque) through various well-known panel types (including the recent "goodies," as some of these panels are considered "disastrous") to the true monocoque type represented by the Cessna. Careful analysis to an extreme we can even think of the true type of fuselage as being those in which the number of cells is greatly reduced. Almost any panel type can be thought of as approaching the cellular structure as the number of stiffeners is increased and the amount of unreinforced shell is decreased.

#### AVIATION December, 1938

could be used as fuselages with complete freedom from local buckling. Since present practice simply abandons around 50 ft. seems obvious that there is a large field for exploration. Remembering that for greatest efficiency the structures should be designed so as to utilize all the mechanical properties of a material it would seem that such a high weight strength factor as 1.00 has been somewhat undervalued in the past. An other view of the problem is obtained by considering that the strength of an Euler column of constant length is only slightly proportional to  $W/B^2$ , where  $W$  is the weight of the column material. Thus, for a given strength the weight will decrease very rapidly as the diameter is increased. Thus, with due care, it is possible to use such a process, such as *monocoque* and *double-skin* construction, etc., but these may not be as important as the use of attachment increases.

#### What the future holds

Steel drivers are fully aware of the practical advantages and disadvantages of most common forms of construction. They are also aware of the need for a local efficiency evaluation on the part of present types. Such an analysis would have to consider features such as local loads, fasteners, joints, cut-out stiffeners, reinforced loadings

and other present features too numerous to mention. Predictions as to what materials are likely to be used most in the future would be somewhat based on a few or other reasons, than the first that one ingests oxygen and readily oxidizes. It is to this oxygen that we must turn our attention, if the efficiency of the aircraft are to be increased. These problems will have to be faced if we hope to build up very large structures on an all-alloy weight-strength basis. Those interested will find the British Air Ministry R. & M. No. 1639, "Stability of a Monocoque in Compression," by J. L. Taylor, to be helpful.

We are on the threshold of a new and better understanding of the structural engineer's problem of transmitting loads from place to place with the least possible structural weight. With such an understanding we can expect to see considerable reductions in "excessive" load, with corresponding increases in payload. And last (but not least) from the engineer's standpoint we can hope for a reduction in the batches which have been brought largely to the market by the increased strength and the rapid growth of the use of plastics, both of which have continued to produce what Dr. Tiedemann has aptly called "a field of enormous opportunity involving the geometrical properties of the more varied construction, combined with the characteristics and other properties of the material."



# Editorials

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## INSULATION

This is not an issue about a living and soldier covering for electrical resistors. It concerns about the blood flowing in the title—or the state of being an island.

Islands may be real, as they may be creatures of the imagination. But in the case of the advantages gained from insulation, the latter is the only answer. The insulation factor is not directly proportional to the insulation offered by the insulation medium—but it is, since, as a state of mind—similar insulation outside resistance.

England, for example, was once an island. Until recently the average ship of war that separates the British Isles from the mainland was a fairly efficient insulation. Because of England has lost some of those fine qualities on the order of European politics were as low as the sea fit. But with the rise of air power the situation has changed completely. As an insulation the twenty old ships of Channel have become about as effective as a strip of wax paper between the terminals of a medium voltage insulation. England is an island no longer—and she is not insulated.

So far, we have been in Europe only long enough to form some ground impressions. And among these is a feeling of dissatisfaction that we live in America as the port of world affairs. Not that we have any special pride in our history or in our achievements. Comparatively speaking, we have no history, and certainly, none a European nation handles most of its social problems for better than ourselves. But we have seen as England with "the wind up" working night and day to build up aerial defense. Before it is too late. We have seen a number of German towns of sufficient size in being put to the torch. We have seen the Spanish Republic's aerial forces march on France with and to restore the Reich to a world power. We have watched the rise of events in Spain and we have learned at close range in the mudholes of the Atlantic River in the east and of the French outfit certain states the White in the west.

Small wonder then that we look with considerable comfort in the two thousand odd ports of course that separate us from Europe. And still, for the present stage of circumstantial development at least, America has taken over England's island position. We are about the only truly insular nation left in the world today, and with both Europe and Asia's affairs require a much greater degree of mobility than this exhibit in the Americas, does not stand.

Seattle, Nov. 10.

## WINTER AT THE AIRPORT

All over the country (excepting Florida and California) snow has appeared, and the airports are closing into the dull winter routine, after one of the best years they have had since 1929 or 1930. Some are closing up their airports and

going into other work. Others are resigning themselves to a cold, less winter, still may be making a serious mistake.

There are those when closing up airports for the winter is the only sensible thing to do. We do not presume that you spend several hundred dollars for snow removal to take up a dozen passengers at a dollar a head

but there is no reason why many of the fields abandoned when the mercury goes below freezing day and night should not remain active during a large part of the winter.

If you need a non-government job to keep down field rates down in winter, you might there have your weekend in prison. Each winter weekend there are hundreds of out-blown youths who do foot in search of winter sports, and are willing to pay for them. Your job is to divert some of that income from the production of chaotic persons, starting riots and similar enterprises. Maybe you will have to flood an owned property in order to get a place for an skating. If you build one, and keep it in order, a series of Sunday night knapsack dances will bring in some revenue if they are effectively promoted. An airplane New Year's Eve celebration could be made to pay big profits. Clubs and church societies are continually seeking new ways to keep members supplied with recreation. Such non-governmental activities serve to focus public attention on the sport, to establish it on the basis of a real, natural recreation season. The rest is easy.

Other special flying areas to nominate the starting or closing seasons. Organize Sunday afternoon and Sunday naps in the support for selected groups. And without any doubt your winter residents will pay you very convenient dividends.

## BUYERS' HOLIDAY

PRODUCTION of airplanes, engines and accessories is going up. Everywhere we look at new highs. One manufacturer is planning to build a thousand airplanes for which he will buy a thousand engines and thousand accessories. Another uses a thousand manufacturers.

When an aviation purchasing agent goes to market to buy anything in large lots he can command some respect from his sources of supply. An equally new schedule of prices is likely to be opened to him. It may not last the friendship of the local distributor since proprietor and a very east has a few nights and weekends of hard labor. But it is sure to re-open.

It follows every man concerned with the purchase of materials in any aviation manufacturing company to work some kind of a weekend going through his stock lists to see where he can save money before some may be purchase of materials in larger quantities.

# Flying Equipment

What's new in aircraft, engines and major accessories

## Bellanca 28-70 "Flash"

New version of Mollison's transatlantic plane to have cantilever wing

THE recent record flight of Captain James A. Mollison and the fascinating New York-Perth race, focus our attention again on the Bellanca Model 28-70 "Flash." A number of refinements have been made in the design and others are planned for ships that may be used in the prospective transatlantic race. During the past summer wind tunnel and static tests have indicated that it will be possible to use the cantilever wing for the heavier types of the present series. With this change and with a "Swing" engine, a top speed in excess of 300 m.p.h. and a range of over 5,000 miles are expected.

The Mollison Flash which crossed the Atlantic at a speed of 227 m.p.h. is powered with a twin-cylinder Pratt & Whitney Wasp Jr. engine rated at 700 h.p., carries 600 gal. of gasoline and is equipped with a non-stop range of about 4,000 miles at a cruising speed of 200 m.p.h. Wings of the Flash are of all wood on the conventional two-spar type with a single spar and all wood leading edge, and false covering. Plywood is also used to cover the top of the leading gear struts and wells. Bracing is accomplished by a system of six rods from the wing spans to fuselage on top and no two specially designed steel supporting struts connected by a sort of sled below the fuselage.

Structure is welded aluminum engine block and tubing, and the landing gear is a single shock strut. The landing gear struts with plywood covering and shock absorber are rubber and are made of welded steel tubing also covered with fabric. The controls on the elevators eliminate the stick-shake adjustment.



Bellanca 28-70 "Flash"

Right picture of the "Flash" showing the landing gear.

Side view showing aircraft sections and basic fixed wing planform.



Three side drawings and equipment arrangement of Bellanca's 28-70 all-metal racing aircraft. Overall fuselage 46'-0"; wing span 20'-0"; height 10'-0"; wing chord 10'-0"; wing span 10'-0"; wing root chord 10'-0"; wing root height 10'-0"; wing area 100 sq. ft.



A conventional hydraulic mechanism retains the landing gear and the locking arrangement is a horizontal sear arm and detent from the retracting device. Retraction is controlled by a lever, from the cockpit, landing loads are distributed over a triangular load arrangement. One leg of each triangle incorporates a spherical shock absorbing strut. Wheels are of the standard Bessis aero-ther type fitted with 3200 pounds tires. Brake blocks with independent hydraulic brakes are also installed. In the retracted position, the landing gear wells are closely covered by tufted cloth covers. A full balloon tail wheel partially enclosed in a streamlined fairing is provided at the fuselage tail post. Landing gear has an extremely wide track which contributes to easy steering.

The transparent material at the cockpit windows is a new development known as "Plexiglas." It can be bent and worked like other conventional plastic materials, yet passes light more freely than standard window glass. A tuft-bladed Hamilton Standard con-

rollable pitch propeller 10 ft. 6 in. diameter is fitted in the rear nose Wang Jenner nacelle. The landing gear is of the most recent A.T.C. type and the most sturdy built metal rear well door. Metal, removable skirt segments are fitted by EDOCS Bessars. Special one shot lubricants for the socket bushes using all pressure from the engine is installed and an oil indicator with an adjustable soap opacity corroborates the oil level. The engine is completely enclosed.

General specifications of the Model 25-30 are: wing span, 46 ft. 2 in.; length, 25 ft. 8 in.; height, 7 ft. 1 in.; wing area, 257 sq. ft.; weight empty, 3,074 lb. useful load, 4,276 lb.; gross weight, 6,250 lb. Performance figures indicate a maximum speed of 130 m.p.h. at 10,000 ft., cruising speed, 240 m.p.h. at an altitude of 4,000 ft., rate of climb to 8,000 ft., 1,560 ft. per min. service ceiling, 26,000 ft.; and absolute ceiling, 27,000 ft.

## Ben Jones S 125

New Mooney powered monoplane for the private pilot

Mooney is approaching but designed to fit the needs of the average private pilot in the Mooney S 125 by Fred C. Mooney, president of Salineville County Aircraft. This newcomer in the private airplane field is a two place low wing monoplane of conventional construction now powered with either the Hispano C4 (125 hp.) or C4S (150 hp.) engine. It is well suited to student instruction and other private use, aviations and has sufficient room for extensive model aircraft storage.

Construction is of the wood, plywood and fabric-covered type using such fabric coverings. The plywood covered, two spar wing is covered with balloon cloth before fairing. Wing and tail unit front are covered with monocoque. The cockpit canopy is built in three interchangeable sections, easily removable. The metal covering at the front of the fuselage is removable to give access to the structure and accessories for inspection and servicing.

Baggage space is provided behind the



Ben Jones Model S 125 (Courtesy)



cockpit, and is reached through an airtight door. Additional protection is given to the rear tail section for maximum engine compartment safety.

All flying controls are on the rear deck level, but space is left for a duplicate set in the front cockpit. Angle lag rods in available both cockpit and the rear cockpit are so designed that they support that part of the tail when the front tail section is lowered.

Control column, rudder and brake control pedals are used and either hand or power is operated individually, with or without simultaneous operation of the rudder pedals. Both bearings are used throughout the control system. Rudder and elevator are controlled by ailerons which also differential motion control consists of a system of cables, bell cranks, and push-pull rods. The trimming tab on the left elevator is operated by a heavy cage wire passing through the elevator hinge corner fair.

The landing gear is of the tailskid type with three-wheel landing gear. The main wheel is on a fork mounted in the shock strut which is attached to a tapered connection to the front gear and to a similar union between the spars. A sliding side connecting the fork and triplex trunnion is incorporated in the landing gear. The height is 7 ft. 3 in. The 8-in. tail wheel has a 360 deg. swivel, a fixed and shock sprung, and a pneumatic discbrake for braking.

Two fuel tanks of 22 gallons capacity each are placed in the wing on either side of the fuselage, and are removable. The fuel tank is located in the rear engine in rubber mounts, the fire wall is strengthened, and the venturi type fueling is attached so as to be free of the main body. These three features contribute largely to the maximum degree of quiet within the cockpit enclosure.

Dimensions are: length, 21 ft. 6 in.; height, 4 ft. 6 in.; width, 8 ft. 6 in.; wing span, 24 ft. 8 in.; weight empty, 1,815 lb.; useful load, 1,235 lb.; gross weight, 3,050 lb.

Performance is as follows:

Mooney C4	Mooney C4S
Wing area	190 sq. ft.
Overall length	21 ft. 6 in.
Width	8 ft. 6 in.
Height	4 ft. 6 in.
Wing span	24 ft. 8 in.
Empty weight	1,815 lb.
Useful load	1,235 lb.
Gross weight	3,050 lb.
Service ceiling	18,000 ft.
Cruising speed	100 m.p.h.
Max. speed	125 m.p.h.
Rate of climb	1,000 ft. per min.
Service ceiling	18,000 ft.
Cruising range	340 miles
Oil capacity	10.5 gal.

## XOZ-1 Gyroplane

Rotary wing craft for Navy tested in Philadelphia

Aeromarine Corp. has just been made by E. Burke Wilford, president, Philadelphia Aircraft Syndicate, of the first flight of the piston-powered gyroplane XoZ-1 built for the Navy Department. The flight was made at Frankford Seaplane Base at Essington, Penn., by Peter Albert Sollner under the direction

of Paul Horpact, flight consultant, Wilkes-Barre, Pa., and Mr. Wilford, that is to say, the XoZ-1 is to be used for military purposes.

All flying and controls are on the rear deck level, but space is left for a duplicate set in the front cockpit. Angle lag rods in available both cockpit and the rear cockpit are so designed that they support that part of the tail when the front tail section is lowered.

Mr. Wilford's organization by

General specifications of the XoZ-1 are: rotor diameter 36 ft. 6 in.; wing span, 25 ft.; nose deck area, 800 sq. ft.; fixed wing area, 180 sq. ft.; engine horsepower (Kinner A-5) 150; gross weight, 2,000 lb.

At the present time the company is

designed to produce the XoZ-1 as an aircraft

which will have a maximum flying speed

of less than 40 m.p.h. and a top speed

of between 130 and 200 m.p.h. with a reasonable takeoff and landing.



The XOZ-1 gyroplane recently tested by Philadelphia Aircraft Syndicate

## W. B. Kinner Coupe

New model resembles folding wing predecessors

A new folding wing, indeed, two place plane has been designed and flown successfully by W. B. Kinner, president of Columbia Aircraft and manufacturer of Seminary-Bellanca airplanes for the last several years. The characteristic folding wings of the new model can be placed in their folded position by one man in about three minutes. Suspense need not be created for the fold underneath operations.

The wing span of the new ship is 40 ft. and the overall length, 24 ft. 2 in. Maximum speed, will be 125 m.p.h. engine, about 225 m.p.h. and climbing about 30,000 ft.

## Wright 3-Way Drive

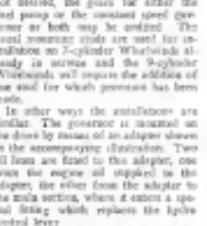
Constant Speed Governor Adapted to Whirlwind

The numerous advantages of the use of the constant speed propeller are already well known, and automatic pitch regulation has been an important contribution in all types of airplane operation. In order to apply the Hamilton Standard constant speed governor to the Whirlwind engine, the Wright Aeromarine Corporation has developed a new auxiliary drive which can be used on all 7 and 9 cylinder Whirlwinds readily equipped with the two position hydro-control mechanism. The constant speed propeller was heretofore available only on the highest powered Cyclone in the Wright line.

Attached below the right magneto at the fast pump drive location, the new

auxiliary drive is a three-cylinder low ratio model of laminated iron from the gear teeth group in the main station.

(Turn to page 40)



Propeller governor installed on a Whirlwind 300. The right-hand auxiliary drive drives the right-side propeller. The left-side auxiliary drive drives the left-side propeller. The constant speed governor pump and regulator, the left-hand unit is constant speed, and the lower unit is fast pump.



W. B. Kinner's new folding wing coupe

# THREE HUNDRED MILLION PASSENGER MILES

OVER ROUTES blazed by Lindbergh . . . the flying Clipper Ships of the Pan American Airways have carried passengers and mail for eight years . . . with an overall record of 99.678% on time.

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## TEXACO *Aviation Products*





# FATIGUELESS FLIGHT



• Metals, as well as men, suffer from fatigue—from loss of strength due to repeated or alternating stress. But metals that are susceptible to this weakness have no place in aircraft engines.

Here are needed special metals, selected for their proven ability to resist fatigue, stress and wear.

To-day the Nickel Alloy Steels are used

for numerous parts of these heavily burdened power plants because their superior mechanical properties minimize the danger of breakage and wear.

Through a partnership with Nickel the steels are made unusually tough and strong, hence more enduring and reliable. Consultation on problems involving use of alloys containing Nickel is invited.

## NICKEL ALLOY STEELS

THE INTERNATIONAL NICKEL COMPANY, INC., 67 WALL ST., NEW YORK, N.Y.

## Buyers' Log Book

What's New in Accessories, Materials, Supplies, and Equipment

### Hydro Degreaser

**Cold Cleaning tank uses Gunk as a solvent.**

**CONSTRUCTION PLATE** is a cold immersion cleaning tank known as the "Hydro-degreaser" are offered without charge by the General Cleaning Company, 3500 Broadway, New York. The property of "Gunk" solvent compound. This device has been developed to remove heavy accumulations of oil and grease from aircraft engines and parts and has found application in many other industries.

The Hydro Degreaser is constructed in two compartments, one for soaking and one for rinsing. In extreme cases, air has sometimes been used to create turbulence in the solvent bath. This reduces the time necessary for cleaning to only a minute or two. The cleaning action is directly connected to the pump which draws off the dissolved oil and carries it in a side outlet. Once the tank has been filled with the solvent, the tank keeps still cleaning and the process continues.

The device eliminates the cost of stoves or gas burners and the solution does not irritate the skin and requires no protection for the operator's hands. It is available in sizes ranging from 10 to 100 gallons. Address: The Gunk Company, 3500 Broadway, New York. **—AVIATION**, December, 1936.

**car** (Philadelphia). The Russell V-6 fast "V-air-speed" is light in weight and only 15 in. x 2 in. x 1 in. in size. It can be mounted on almost any elevator cord up to 220 volts and under 600 watts input. To obtain 1,000 hours life at contact points it should not ride over a maximum of 100 amperes at 80 volts. The device is a condenser across the contacts to increase the life of the device. Wiring is completely enclosed.

The Vair-Speed is operated by turning a short switch on the case, giving a range from 1-4 rated speed. There are no moving parts in the case and air from the motor fan dissipates the governing medium—**AVIATION**, December, 1936.

### Thor Spray Gun

**Model 7 has 20 different nozzle tips.**

A more accurate and capable of spraying materials ranging from airplane dope, lacquers, and enamel up to liquid desiccants and can-kuck asphalt has been introduced by the Metal Finishing Company, New York. The Thor Model 7 can be provided with approximately twenty different nozzle sets, all of which are interchangeable. Among the improvements claimed by the manufacturer are greater speed, less air consumption, and more accurate atomization.

Using the Thor spray gun, the nozzle is inserted into the gun body which is enclosed in the trigger. The body is of drop forged aluminum with a black electrolytic coating for surface protection—**AVIATION**, December, 1936.

### Jack-of-All-Trades

**American Chain Company offers stretching Jack with scissor-jack**

For operations involving stretching, pulling, loading or lifting, the Welded Chain Division of the American Chain Company, 1400 Broadway, New York, offers a stretching jack of unusual design. The 34-ton handles stretching loads to 4,800 lb.

What is the stretching device that has over teeth in a spooler wheel mounted in a frame. Raising and lowering a handle attached to an eccentric axle transmits power to the shaft through two gears—**AVIATION**, December, 1936.

### Vari-Speed Control

**For portable electrical tools and other small motors**

A motor control for portable electric tools and other small universal type motors has been placed on the market by the Pennsylvania Aircraft Syndicate



Russell Vair-Speed.

### Tubing Data Book

**Sunbeam publishes loose leaf handbook for engineers**

An extremely useful handbook of aircraft tubing has just been published by the Sunbeam Tubing Company (Bridgeport, Pa.). In the 39 pages of tables, charts, and formulas are tabulated all the data necessary for accurate, square, and elliptical sections. Elliptical strength curves for circular, rectangular, and square sections are tabulated. Yield points higher than commonly employed in present construction of structures in aircrafts. Other data values, such as modulus of elasticity, are also given in the loose-leaf pages which may be replaced or increased in number as the occasion arises—**AVIATION**, December, 1936.



The Thor Model 7 spray gun.

### Dalco Landing Light

**Hydraulic Retractable Landing Light introduced on Coast**

In the announcement and description of the Dalco retractable landing light in the November issue of **AVIATION**, a typographical error was made in the first word of the name of the manufacturer. The above spelling is correct—**AVIATION**, December, 1936.





# HAMILTON STANDARD Controllables

ON  
NEW  
BASIC  
TRAINERS



To request Army student pilots with an appreciation of the performance of high-performance aircraft, the new basic training days of North America and Beverly are now equipped with Hamilton Standard Controllable Pitch Propellers — further evidence of the widespread use of the controllable pitch principle.

**HAMILTON STANDARD PROPELLERS**  
EAST HARTFORD, CONNECTICUT  
DIVISION OF UNITED AIRCRAFT CORPORATION

## AVIATION December, 1938

company had its usual contract in the general construction, and, designated as Pennsylvania Airlines & Transport Co., continued its operations to Detroit, where it is now based.

Pennsylvania was succeeded however, by the newly-formed Central Airlines, which has held the Washington-Detroit contract (Route 11) until the recent consolidation of interests.

Meanwhile Pennsylvania bought the Toledo, Ohio, franchise, which it now possesses, the Detroit-Wichita route, Route 10, Lake Michigan, in 1936. This enabled the company to make a useful bid for the contract for this route. Passenger and express service on the Washington-Detroit route has continued for the past two years on fairly comparable terms.

Officers of the new corporation are drawn almost entirely from Pennsylvania's list. C. Bedell Mingo will continue as president, Frederick B. Crawford, vice-president and secretary, W. J. Austin, vice-president in charge of traffic, E. G. Howell, treasurer. Only members of J. H. Cullen, assistant secretary and assistant treasurer.

## Production, Sales

### Stearman for Brazil. United Aircraft increases wages

Tom Stearman, Army Aviation Department, called a meeting last week in October with the Stearman Aircraft Company, of Wichita, Kansas. The contract calls for 20 Stearman Model 70-C3 advanced trainers. Power will be from Wright Whirlwinds of 420 hp. The price is \$90,000, and the first set of five planes will be delivered next March.

One of Lockheed's new Model 10s (see Aviation for September, 1938) which won the Bureau of Air Commerce competition for a small two-engine transport for feeding-in service, has been delivered to the Bureau of Air Commerce for the use of Department personnel. The plane cruises at 212 m.p.h. on the power from two Pratt & Whitney Wasp Juniors. It is equipped with a Spruce Gyroplane, Western Electric two-way radio, and a new Lockheed Products horizontal stabilizer in a trimaran profile and attached to the underside of the fuselage directly ahead of the engine. Delco equipment is installed on the leading edge of the wing surfaces. Fuel storage for a 1,000 mile range is provided.

Fairchild Aircraft Ltd., of Canada, has sold 20,000 shares of capital stock to Allied & Co. Ltd., of Montreal, at \$5 a share. The stock was revalued at \$8 a share. Proceeds will be used for working capital to take care of the 50 per cent increase in costs for the year 1939.

As a result of the sale, Fairchild Engine & Airplane Co., which was

unable Dec. 12, to make delivery of record Dec. 1, On the basis of 2,365,763 shares outstanding, this would mean a \$1,221,000 dividend requirement. The New York-New England distribution for the 1937 series at Ryan Aircraft Co. plant in San Diego, has been granted to D. C. Whitney, Inc., North Beach Airport, New York. Delivery of the first West Coast factory, Whitney ordered four planes, one of which were for immediate delivery.

Ryan recently decided to finance plane sales on the basis of one third purchase price down, and the balance as twelve monthly installments.

The Ex-Cat-D Aircraft & Tool Co., Detroit, Mich., has adopted a group insurance program providing approximately 800 employees with sickness and accident benefits, and a total of nearly \$900,000 of accidental death and dismemberment protection.

On Dec. 1, the Aircraft Co. of Bradford, Pa., has announced a \$100 reduction in the price of the Taylor "Cub" bringing the price down to \$1,270. W. S. Piper, general manager of the Taylor Aircraft Co., has estimated that the plant has a stock of about 300 Taylor "Cubs" at 40 per cent.

Effective November 17, all divisions of United Aircraft Corporation

granted a flat wage increase of five cents an hour for their hourly wage employees. This will add more than \$100,000 to the corporation's annual payroll. The Board of Directors also voted a 50 cents per share dividend.

The Stearman-Houston Aircraft Corp., which has been expanded to manufacture the Houston Y pusher developed for the Bureau of Air Commerce light glider competition, has registered 280,000 shares of common



LYCOMING STINSON

The maintenance work on the aircraft, Rev. Paul Schmitz, stands at the front of the plane he has designed as the successor to the long-time competitor of Cessna.

was issued in October to expand the airplane and engine manufacturing activities of Fairchild Aviation Corp. Since its creation and recent success, will have no stock interest in the Cessna company, resulting in about 30 per cent of 40 per cent.

Effective November 17, all divisions of United Aircraft Corporation granted a flat wage increase of five cents an hour for their hourly wage employees. This will add more than \$100,000 to the corporation's annual payroll. The Board of Directors also voted a 50 cents per share dividend.



COL. LINDBERGH  
discusses his new B-10 Bomber with its designer, President George Miller. The plane was dedicated to the Collier competition, and is powered with a 300 hp supercharged Mercury twelve-cylinder (20-10).



BRAZILIAN PILOTS WILL TRAIN

In these Brazilian Model 24-15 advanced trainers, 20 of which have been ordered, the Boeing Aircraft Company will train Brazilian air corps pilots.

stock on the San Francisco Carb exchange. The company is headed by Lloyd Steersman, and has constructed a new plant adjoining the San Francisco Municipal Airport. Fifteen Vs have been constructed for the Bureau of Air Commerce.

Walter Eddle & Co., Inc., has developed a new improved valve for low airplanes by extrapolating curves showing a weight reduction of approximately 12 per cent. Weight of single-wire faced control valves is now down to about 15 lbs.

The Aeroflot, Soviet Air Transport Corp., headed by W. E. Konev, is planning construction of a factory adjoining the Leningrad Batai Municipal Airport. It is planned to put into production a two-place low-wing monoplane powered with up to 1,000-hp. (one 500-hp. and one 500-hp.). The schedule calls for an initial program of two planes a month.

Grumman Aircraft Engineering Corporation, of Farmingdale, N. Y., has acquired two-sections of rights to the license production of the Grumman G-14. The gear is the type in which the main strength arm consists of two legs joined by an offset stiff locking hinge, and one leg of which is a shock absorber.

Major Alexander P. de Seversky, president of the Seversky Aircraft Corp., announced in October the purchase of the plant which the Ranger Engineering Corp. was leaving from the Aviation Corp. Shipyards in Stamford (Bantam) during Oct., stating that Ranger had sold the property only if it intended to expand. There for a while, the master was deadlocked. Then

last month the two companies announced a settlement of the dispute, which will buy from Aviation Corp. its present plant, and will use the adjacent field jointly.

The Detroit Car & Foundry Co., which has been manufacturing and repairing export license elevators at the Grumman F7F two-seater fighters, will also handle export manufacture and sales of Vincent Barnett's transport monoplanes. Sales efforts will be made in Canada, England and South America. Last month a 15-plane-a-day Avro Arrow gave the plant a capacity of fifteen planes a day. That should have read fifteen planes a week, as it was in the accompanying story.

## Face Enough

**Two lines join TWA in cutting prices.**

Is not last year, passenger fares of all U. S. airlines have been down more than 50 per cent—down 12 cents in 1938 to 57 cents in 1939? This has been the result of growing traffic volume, more efficient equipment and ever-increasing competition. For the past four years, there have been or have been threatened, the movement down and down has been less than half a cent since.

Then Sept. 26 TWA announced dramatic fare reductions as an inducement to traffic through the winter months when it usually slack off. The TWA philosophy is that it's more economical to fill these planes 100 per passenger than to operate with a lower load factor because of higher fares.

About simultaneously after the TWA announcement there was talk of a

surprising rate war among the airlines. But not after another, the other three of the "Big Four," American, Eastern and United, disclosed all thought of participation in the war. Eddie Eichendorff, general manager at Kansas City, said, "There is no such thing as 'hot' side air transportation. For ten years now air lines . . . have fought courageously . . . to place air transportation at the economy where it is today. . . . When we are in a position to show black instead of red on their books, this is the time for these men who have stuck to their posts during that long, lean year struggle, to get their rewards. . . . To indulge now in a childish price war would be to knock the proper weight out of travel, the latter having turned which we have been so laboriously reaching for ten years."

A few days ago TWA's announcement of fare cuts, Chicago & Southern Air Lines announced "substantial fare reductions" in the trip with the recent opening. TWA's Carl S. Lindbergh, C & S president, believed that "the fare has passed for regarding air travel as a service available only to the few." We believe the increased traffic volume which will result will not only increase company revenues, but will widen the audience to air travel.

Third line to reduce fares was Braniff Airways, which instituted lower winter rates effective until March 1 of next year.

Meanwhile, before Braniff came up the private, C. R. Smith, energetic president of American Airlines, also decided to get into the act. His comment was, "Members of the industry figure 80 per cent of the passenger miles of the country are not paying out rates one effect." But he added, "We will watch with interest how the public likes our rates as far as transportation."

W. P. Farley, president of United Air Lines, held that "United is re-enforcing the public's approval of its present rates, salaries and service, which have enabled us, without cutting rates, to operate at the highest load factor in its history."

## Calendar

Sept. 26-27—North American AFM Association, Minneapolis, Minnesota, Air and Space Show.

Sept. 26-Oct. 4—1938-Matador Aircraft Show, Grand Central Palace, New York.

Sept. 28-Oct. 19—National Aircraft Parts Show, Hotel Plaza, Los Angeles, Calif.

Sept. 28-Oct. 19—National Aviation Show, Los Angeles, Calif.

Sept. 28-Oct. 19—First International Show of Applied Mechanics, Los Angeles, Calif.

## NEW WINGS FOR DEFENSE . . .

*The Boeing  
YB-17*

★ Pioneering is a Boeing tradition. This is the first of a series of giant four-engined bombers now being built for the United States Army Air Corps, a type of plane which is establishing a new trend in commercial as well as military designs.



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## Miami Cruise

Two eight-plane groups planning same flights to annual maneuvers

A general feature of the Sixth Annual All American Air Meet, to be held at Miami, Fla., on Dec. 11, 12, and 13, will be the arrival at the seaport airport of two large groups of light planes. The Taylor Aircraft Co., of Brookfield, Wis., is sponsoring a mass flight of Taylor Cub, Taylor Aerocar, and Taylor Air Chief planes. To date 280 planes are signed up. The Taylor Company will pay each member's round trip expenses for the fliers who will be in Miami. The "Cub Country" is under the direction of Ted Wild, of Taylor, May 12. Tom Nease, Fred, and Ralph Lockwood, Bell Off Co.

A normal group of planes, which may range in size from 10 to 50, will be made up of Arrows, and will leave Cincinnati, O., Dec. 10. The Arrow Corporation of America, which is sponsoring the event, will supply planes to fliers who may stop en route to the cruise through the Midwest. Every Arrows owner east of the Mississippi has been invited to pass.

## Procurement

Army orders 267 planes, 66 Consolidated VPBs for Navy

The Army Oct. 26 announced award of a contract for 117 basic trainers to the Standard Aircraft Company, Inc., Indianapolis, Ind. Contract price is \$1,452,600. The planes will be denoted locally as the Ar Corp. Primary Training School at Randolph Field, Tex. The Ar Corp. said that "although somewhat slower, this plane is very similar in appearance and general flying characteristics to the current primary training planes. At the same time has the ease of maintenance, combined with sterility of construction which is required in a training plane."

Get 26 more another basic trainer order to the Standard Aircraft Company, Inc., Indianapolis, Ind. at a price of about \$450,000.

A second large order to North American came Nov. 19, and called for 120 observation planes. The planes are three-place, including canopy with removable leading glass and tail wheel. They will be built with single 450-hp. Wright Cyclone engines and will be equipped with training-type flaps. Purchase price is \$3,425,000. Secretary of War Harry H. Woodring said, in announcing the award, that the contract generally completed the procurement program under the 1936-37 appropriation. By the end of the year, he expected to complete a program for the acquisition of 300 planes.

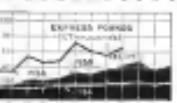
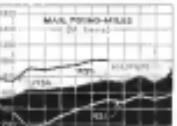
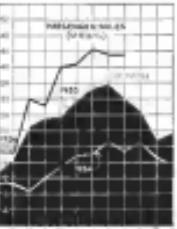
Consolidated Aircraft Corp., San Diego, Calif., was awarded a \$6,096,900 contract for 100 bombers for the Navy. Price, \$61,800.

The Coast Guard has asked for 150 planes. From two to six passengers, convertible for use either on land or water.

In a circular dated Oct. 30, the Air Corps asked for bids on 46,360,000 gal. of gasoline, as annual storage specification.

## Traffic

Latest available statistics from the Bureau of Air Commerce and the Post Office Department—Domestic airlines only



Be the first to believe in this flying fortress. It is the best of flying in speed from the ground up. Flying assembly starts at 800 ft., and is attained just before take-off. Max. altitude is 30,000 ft. It is the same 300 design that flew from Seattle to Durban last year at an average speed of the 300 ft. It is powered with four 1,000-hp. Wright C. Cyclones.

Over 2½ Miles

OF ROEBLING CABLES  
FOR THIS NEW BOEING BOMBER



Using 1/8" Dacron  
for U.S. Army Air  
Corps

HERE are 2,265 ft. of Roebling Control Cables and 11,617 ft. of Roebling Electrical Cable used on each of the new Boeing YB-17 Bombers. Thousands of these planes—America's largest land plane and the world's famous bomber—will soon be delivered to the U.S. Army Air Corps. Because they assure the utmost of safety and durability, Roebling Aircraft Cables were the choice for this outstanding fleet as they are for a big majority of planes built in this country.

Roebling Wire Aircraft Products include...  
*Tinned Aircraft Wire, 19-wire Aircraft  
Strand, Tinned or Galvanized Aircraft Cord  
(635, 785, 7895), Tinned and Galvan-  
ized *Terrene* and *Thimbles* Sealing and  
Locking Wires, Control Strand and Casing  
Power and Lighting Cables, Welding Wires*

JOHN A. ROEBLING'S SONS COMPANY  
TRENTON, N.J. Branches in Principal Cities



ONLY A FINE PRODUCT MAY BEAR THE NAME ROEBLING

AVIATION  
December, 1945

Office Department for a second hearing in negotiation for the purchase of the TWA. The first application for purchase was denied. But the British officials have been encouraged to make better overtures with Fox American at Brewsterfield and TWA at Kansas City.

Edithson has been selected as the northern division of Fox American. Areas for use are the same. A trans-Atlantic survey flight which is scheduled to start next spring. The PAA, trust, for its area of the Municipal Airport, runs for an initial period of ten years, with renewal privilege for three additional five year periods.

Edithson is being considered as possible as terminals—notably Charleston, S. C., and Fort Washington, N. Y.

Haward Airlines is planning a series of light busses on the route from Chicago to the Twin Cities and from Sioux City to Sioux City—Lockheed Constellation with the route on the Sioux City—Bismarck route.

J. R. Walker, TWA, was president in charge of traffic, has reported that over the first half of this year each passenger flew an average of slightly over 800 mi., and TWA planes carried over 800,000 passengers. The new Boeing 747s showed a total of 64,025 revenue passengers, compared with 41,745 for the same period last year.

The first nose section of 2026 brought a new record for Haward Airlines, with traffic gains from 35 per cent to 37 per cent for passengers, staff and express

**Lockheeds Abroad**

British Airways orders four Avro 681 aircraft ships for Stockholm run

Paramount owners of British purchases of American military planes have been flying across the Atlantic since early October. Great Britain's planes are known to be loaded up with Air Ministry orders in a frantic rush to get up the first few planes for post-war Germany and France. On the side of the coin on continuations of the rumors could be had, the companies mentioned for possible orders indicating that they were interested, but had heard nothing from Britain.

There is information that British Airways, Ltd., had ordered four Lockheed Electra jet commercial sites. This order was made necessary because the Air Ministry issued use of De Havilland 88A's for night service on the London Stockholm run, and the state of the British industry left no percent necessary changes in time for inauguration of service.

The Lockheed will be standard 30-passenger Electra with several airline

equipped. They will be powered with two Pratt & Whitney Wasp Jr. engines, will be equipped with Lockheed standard pressurized. Delivery date was not announced, nor was price revealed.



"TIME FLIES"  
A baby companion for French Biscuit's new steel ship "Time Flies". It is a Vought  
Corsair, and here the baby takes delivery of the French steel.

time for the third quarter of 1946 ended Sept. 30 was \$38,278. Net loss for the first nine months of the year was \$13,389.

The Curtiss-Wright Corporation and subsidiaries have reported net profits for the first nine months of 1946 of \$2,261,131. There was a \$400,000 loss in the first quarter. With the exception of the Curtiss Wright engine subsidiary, showed a profit of \$499,536. For the September quarter, profit was \$200,915.

New issuance for the Glass L. Martin Company netted \$4,200,000. Total assets of the aircraft and aircraft parts division of the company for Sept. 30 were estimated at \$344,000, with gross sales this year at more than \$5,000,000.

During 22-4 cents a share on \$22-12 capital shares outstanding, TWA has released a net income of \$12,800. This is the largest in the company's history. Dividends will affect stockholders the right to subscribe to \$27.211 additional shares at the rate of one third share for each share now held. It is expected that this offering will yield between \$2,500,000 and \$2,900,000, most of which will go for the purchase of new aircraft.

United Aircraft Corporation has reported a net profit of \$202,248, or 28 cents a share, for the first nine months of 1946. This compares with \$400,213 for the same period last year. For the quarter ended Sept. 30, net profit was \$36,185.

Em-Ci-O Aircraft & Tool showed a net profit of \$351,387. In the quarter ended Sept. 30.

For the nine months ended Sept. 30, American Airlines reported a net loss of \$396,832 after taxes, depreciation, and other deductions.

United Air Lines showed a net revenue of \$367,887 for the third quarter of 1946.

Waco Aircraft Company showed for the first nine months of 1946 a net loss of \$40,142 against \$11,527 loss a year before. Sales \$830,081 against \$830,532.

**...Furnishing the dependable indications essential for transport service**

Model 404 Cylinder Temperature Indicator

Model 405 Carburetor Air Temperature Indicator

Model 406 Tachometer Indicator

Model 408 Oil Temperature Indicator

Model 504 Volt Ammeter

Model 602 Air Temperature Indicator

Model 603 Cylinder Temperature Indicator

Model 604 Radio Compass Indicator

Model 605 Ammeter

Model 606 Fuel Level Indicator

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Aircraft Instruments

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## Schools, Services, and Airports

A state-by-state tour of the flying fields

• **ALABAMA**—James Endall, commercial operator at the Birmingham Municipal Airport, is flying with American Airlines as co-pilot between Memphis and Newark. The Auburn Aero Club, organized at the beginning of the present term, has a membership of 25 students. A Taylor Cub has been purchased by the club. The first annual pilot flight meeting at Birmingham, is in progress for the day. James Brown is director. Flown at the **ALABAMA AIRPORT**. The Birmingham Aero Club had its monthly meeting and lecture October 8 at the Tinsley Hotel, Birmingham. Jim Martin, the pilot director, has arranged for the annual pilot flight meeting to be held at the Concourse Airport here in October.

Local clubs are planning the dedication of the Lake Black Municipal Airport, in which case more than \$200,000 has been spent in the past eighteen months. Work on the Autar Airport's improvement project is scheduled for completion by the WPA early in October. About \$70,000 has been appropriated for construction of the cabin, engine, radiator, fueling, drainage facilities and completion of the administration building and hangars. Work was started early in October on a new \$90,000 airport at Jauns.

• **ARKANSAS**—The Arkansas Flying Club is planning a recreation of the club meeting room at Pleasant Sky Inn. Membership in the club has grown to thirty. The newest members are W. M. and Charles A. Tweed and J. Semmens. New wings for the rear day lights to the Taylor Cub of W. W. Williams have been under the direction of R. M. Semmens. Thirty thousand feet of cable is being used.

• **ARKANSAS**—The Free State Flying Club has endorsed a proposal for a bond issue for the purchase and improvement of Pine Smith Airport.

• **CALIFORNIA**—The first air show in several years was staged at Sacramento Municipal Airport on October 6, sponsored by the Sacramento Junior Chamber of Commerce, the local chapter of the Motion Picture Aviators Association. Participation was primarily by local commercial and private pilots and the well represented programs and good publicity brought out one of the biggest crowds since the dedication, estimated at close to 25,000 persons. The program included a parachute jump by Gandy Raugh, stunts, a bomb drop-

from a biplane, and coast fly-by. C. T. Jones, who operates a crop dusting service from Sacramento Airport, reports the purchase of a Challenger Travelair from David Weston of Woodland, and the relocation of his business.

• **CONNECTICUT**—The fifth annual air show commemorating the anniversary of the opening of the New Haven Municipal Airport was held Oct. 12. Over 50,000 spectators were present. Performers included Fred Brooks, Louis de Poerier, John G. Johnson, and Ted Lockhart. A crowd of 18,000 watched the Bantamian ceremonial air show at Melrose Airport, Stamford, Oct. 18. The show was sponsored by the Bantamian Flying Club under the direction of De Joseph L. Levy and James Maloney. Thomas Bay and Constant Mueller of New Haven won the crowd. George Adams of Boston was the 35-mile race.

• **DISTRICT OF COLUMBIA**—Construction of approximately 100,000 square of paved runway, two strip and asphalt at Washington Airport is now under way. At a half-day open government session during the construction of the new airport, the public have installed a road system of radio control for all landings and take-offs. It is considered possible that the system may be made permanent. Zoned charts will be shown to pilots using the port regulator. Pilots are required



**TAYLORCRAFT**  
Left: Kenneth Nichols (left) production manager, and E. G. Taylor, president of Taylorcraft, display the model of the instrument in the light plane field.

to contact the control tower before landing. The tower indicates who is available for landing. The control is in operation 24 hours a day.

\* **FLORIDA**—The field lighting installed by the WPA at Fort Pierce Airport was ready for use Sept. 22. The runway is furnished by the city and the three long runways are now lighted every night from sunset to sunrise. A new 1500 ft. control tower was completed September 29. The 1000 ft. long runway has authority for landing at the Cypress Wright Flying Service hangar at Miami Municipal Airport. Purchase price is set at \$20,000.

\* **INDIANA**—A new steel hangar at the South Bend Plant Club early in October served the club a profit of \$1250. A \$14,000 WPA project is under way at Rochester Municipal Airport. Plan call for an 800'x100' ft. hangar with a 20x100 ft. lounge across the north end. The hangar is estimated to cost \$1000 per foot of the first floor. An air strip was authorized at the Penneworth Airport last in October.

\* **ILLINOIS**—The Marion Chapter of the American Legion has authorized \$200,000 bond issue to construct the Dan Marion Municipal Airport. The local post of the American Legion has endorsed the plan. A WPA project under way at Iowa City will provide reconstruction of the two present runways and construction of a third runway at an approximate cost of \$90,000.

\* **KANSAS**—More than 20,000 persons witnessed the air show at Wichita Airport last in October. The show was

at \$40,000 has been approved for construction of landing strips and a hangar at the University of Idaho, Moscow, Idaho. POCATILLO

AVIATION  
December, 1936

\* **ILLINOIS**—An air marker 300 ft. high has been completed by WPA workers at Marion Airport. The north-south-numbered runway at Marion Municipal Airport was finished Sept. 16. Presidential approved has been given to a \$172,000 WPA project for the improvement of the Milwaukee Municipal Airport. A WPA project that cost \$1000 complete attention air show held early in October at the Marion Airport. The show was managed by John Bryan.

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AVIATION  
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# PLEXIGLAS

## Crosses the Atlantic



PLEXIGLAS was used exclusively in the cockpit hood cover of this Bellanca Model 28-70 "Fleet," which recently set a new record for an eastern crossing of the Atlantic Ocean. Plexiglas was chosen for this cockpit cover due to its strength, remarkable transparency, resistance to weathering and ability to be shaped into two- and three-dimensional curves conforming exactly with the curvature of the streamlined body of the plane.

Plexiglas is a tough, colorless, transparent, synthetic resin of low specific gravity, which lends itself readily to forming into curved plates suitable for windshields, windows and cockpit covers for aircraft. This material is available for use in aircraft in plates and sheets about as large as 36" x 48" in thicknesses from .030" to .250".



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T. CLAUDE RYAN

director and big enthusiast Ryan & Ryan Co., New York, who will use it for research work. The 4000 ft. runway is to be 1000 ft. wide.

The blind Umpiring Board has been formed at Cananda, where it has leased the Cananda Airport from Mike Clegg. See: Yergerman, a president and Ernest French is treasurer.



**BENDIX**  
  
**HT**  
**SPARK PLUGS**

FOR  
 AIRCRAFT  
 ENGINES



**SCINTILLA  
 MAGNETO CO.**  
 Subsidiary of Bendix  
 Aviation Corporation  
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 IN AIRCRAFT DEVELOPMENT**

If YOUR own engineers are tied up with present problems, and development work is lagging, we are in a position to work with you, promptly, and at minimum expense. Qualified service on research, patents, management and development engineering. Inquiries invited.

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 ■ SQUARE  
 ■ ROUND

Made in accordance with A. S. T. M. specifications to meet all present requirements for strength, weight, and durability. Each tie rod is turned from a single piece of steel and heat treated to obtain maximum strength and minimum weight.

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 Makers of Tie Rods, Sprocket and Cable  
 used by all leading aircraft manufacturers  
 and engine experts.



**AVIATION**  
 December, 1936

set up by Peter Dunn of Holderness in a Rye, and Arthur Harris of Clarence as an Eagle. The air meet was sponsored by Howard Hartman, manager of the field.

• NEW YORK — The southeastern members of the Scarsdale Hills Airport, Scarsdale, Rye, had a field opened 400 ft. from Standard Flying Service, Inc., has recently sold a Taylor Cub to E. C. Gandy of Chester and Fred Hirschberg of South Orange. . . . The ATLANTIC CITY Flying Club is planning a flying visit to New York to enjoy a meet made by members of New York's Tidwell Club.

• NEW YORK — Work was started late in October on a new hangar at the Staten Airport. The WPA has furnished about \$17,000 and the village will contribute about \$12,000. The hangar will be located near the present garage, to the right of the gate. It will be 80x60 ft. The previous work at Floyd Bennett Field on Miller Hill, GLEN COE, was scheduled to start late in October. The Glens Falls Chamber of Commerce has purchased the field and will have it open to the public. The Glens Falls Airport has been leased by Capt. H. J. Barnes, who will operate a charter service and flying school. Associated with Barnes will be A. K. Adelby. Barnes now has over 20 active students. Albert G. Wingate, Lawrence C. Archer, and Randolph A. Hoback, of the Amsterdam Flying Service, will purchase a field at Halcott Station, where they will establish a shop. . . . According to William Irving, the management of the B-W Airport in Letitia will go to Dr. Harold Trout and April Trout will release the field from the present manager, Capt. Russell W. Holderness.

• NORTH CAROLINA — By early October plans had been practically completed for leasing the (contested) airport at REEDVILLE. Capt. D. J. Driskill, for a period of one year, Driskill will have exclusive use of the field for commercial purposes, but the lease will not interfere with or prevent private use of the airport.

• OHIO — Airways, Inc., of CLEVELAND, has purchased the first tower from Tamm, Mitchell, contractor at Tamm-Cleveland Airport October 1. The AIRPORT PORT CLINTON, has made a deal with the city of Port Clinton, which will be held at the FLYING Airport early in October. Lt. Joseph C. Barker gave an exhibition of stunts flying 15 pilots from CINCINNATI, DAYTON, MIDDLETOWN and CINCINNATI, KY., and part in a Syria cruise at Middletown Airport Oct. 12. The cruise was

sponsored by the Queen City Flying Service, Inc. George Wolden, manager of Middletown Airport, was host to the group.

• OKLAHOMA — The Hill Brothers Flying Service has been established at Lawton Airport. It is run by Wiley and Mollers Hill. Flying equipment is a Curtiss Robin. . . . The Harry Young Flying Service, OKLAHOMA CITY, has purchased a new plane, Waco Oklahoma chapter of the National Aeromodel Association, were present a convention at Tulsa, November 3-5.

• OREGON — Flying work has been completed on the two 1,000-ft. runways at the Portland Municipal Airport.

• PENNSYLVANIA — Clarence is considering establishment of an airport on the Laundry Farm, 4 miles east of the town. The Foreign Council approves the plan. WPA licensure will be sought for the site's improvement. . . . WPA has approved an allocation of \$100,000 for construction of an airfield at Monaca, PA., which will be equipped in the runways at the WHEATLAND Municipal Airport at WHEATLAND was started late in October. The project is scheduled for completion in December 15. A second project calling for erection of a hangar and construction of a lighting system has been applied for, but is still awaiting WPA approval.

• RHODE ISLAND — Wilard M. Fletcher, Chief of the State Division of Airlines, is planning a system of five regional airports throughout the State. The first will be at MORGAN BEACH, NEWPORT, N. R. . . . Weymouth and Newbury are being considered. Weymouth recently turned over 8,180 acres out of land to the State.

• UTAH — Work was scheduled to start late in October at the SALT LAKE CITY regional field airport improvement project. The Salt Lake County will make \$75,000 in spontaneous contributions. The airport improvement project at BRYCE CANYON was scheduled to start

• SOUTH CAROLINA — A record established at 8,000 feet Johnny Crossell was the first-all-race which clinched a two-day meeting of the Carolina Aero Club at CHARLESTON, October 18. The winner was John Crossell, who won the short racing contest. E. P. Farris, manager of the CALIFORNIA AIRPORT, was the light plane race in an Autocar.

• SOUTH DAKOTA — Work on WATERTON's new \$67,000 airport, which is to be completed in a WPA project, is expected to be completed this fall. Construction on the new administration building is finished.

• TENNESSEE — Memphis is seeking WPA assistance for installation of a field lighting system for Memphis Municipal Airport. The estimated cost is \$3,500. The project has been submitted to WPA. . . . Memphis was planning a debt-bond issue Nov. 1, signifying completion of work on the new municipal airport.

• TEXAS — Work has started on a WPA runway project at BROWNSVILLE, a project to cost \$45,000. WPA will provide \$35,000 of the total. Plans call for two runways, one 2,500 ft. long running north and south and another 1,500 ft. running northeast and southwest. . . . Construction work on the new runways at GALVESTON Municipal Airport has been completed. Another project has been submitted to the WPA a request for a \$20,000 grant for improvement work at Ellington Airport.

• UTAH — Work was scheduled to start late in October at the SALT LAKE CITY regional field airport improvement project. The Salt Lake County will make \$75,000 in spontaneous contributions. The airport improvement project at BRYCE CANYON was scheduled to start



**DOUGLAS B-18**  
 Flying into Irving, transported by students of the Cincinnati-Wright Field Institute, Ohio. The big load carries 32 passengers.

early in October. The project is being sponsored by **Laurens County** as a WPA project and will cost about \$30,000. Plans call for an 800-ft. hangar and two 5,000-ft. runways.

• **VEGAS-MONT**—Nearly 8,000 spectators attended the dedication ceremony of the **Las Vegas-Montgomery** Airport early in October. The field was dedicated by the WPA as a joint project by the two cities. It cost about \$60,000. Two new runways were constructed. The east-west runway at the **Las Vegas-Montgomery** Airport is being extended from 2,500 ft. to 3,000 ft.

• **VEGAS-MONT**—An air show, attended by 2,500, was held at **Pearson Glass Airport**, Las Vegas, late in October. Bill McCullough, flying a Taylor Cub, won the high flying contest.

An airport will be established on a 40-acre tract two miles east of **Blackman**. Two 2,800 ft. runways will be constructed.

• **WASHINGTON**—There is nothing in \$60,000 WPA appropriations for improvement at the **Franklin County Airport**. Plans call for grading and leveling and construction of two runways, one 4,600 ft. long and the other 3,600 ft. long.

## Schools

• **PARK AIR COLLEGE** at **Park Airport**, East St. Louis (Ill.) has been given an assignment to develop courses and school equipment. It is generously illustrated with pictures of the shops and the classrooms.

• Because of the large enrollment in the airplane and engine maintenance night school, **Aviation Schools**, St. Louis (Mo.), announced, through Manager Joseph W. Weller, start of another such class on October 19. The course deals with the theory of flight, aerodynamics, and maintenance and overhaul of planes and engines.

• Expanding all facilities of the ground school department at the **Ryan Schools**, or **Aeromotors**, **Lindbergh Field**, San Diego (Cal.) in order to accommodate the rapid increase in student enrollment, President T. Claude Ryan has announced the following: Virgil McCallum, airplane instructor, and Marvin Wessling, assistant engine instructor. For the second term in the past two months, the Ryan School has been unable to meet the demands from the aircraft industry for trained men. One recent report called for at least 70 Ryan graduates.

• One of the features of the short course in aeronautics which will be part of the program of the School of Adult Education, conducted by the General Extension Division at the University of Florida at **Orlando**, will be an orientation of the students to the field of aviation for a period of 12 weeks. University liaison student enrollment is encouraged. Mr. Raymond, at his present age, has no time to deal with it. His tenure as head of the school appeared to be that it would not be good practice to reduce study enrollment too soon, but Prof. E. G. Reed, of Stanford, in discussions regarding such difficulties and demanded the closest approach to absolute silence.

For the avoidance of rough air disturbances, Mr. Raymond looked only to flight at high altitudes and required the supply of oxygen to the aircraft, or the cabin, or to the cockpit or to and to any that was there and as yet no one could advise without strumming at the moment to determine a solution. There was much to do, in airplane structure, in power plant development, and in maintenance, mechanical detail, before "interceptor" airplanes would emerge from the conversational stage. At one study group, for instance, Prof. Clark had purchased a new Taylor Cub for use as flying laboratory at the **Ulysses (Mo.) Airport**. Roger Dean, Ray, has organized a flying school, to be known as **Capitol City Airways**, at **Lansing (Mich.)**. Clayton Joseph, in manager of the **Michigan City** Flying Club, a chapter of **Stearman Flyers**, **Flint (Mich.)**, has announced an 18-week course of ground school instruction. **Camp Stanley** has purchased a Taylor Cub in which to give flying instruction at the **Engle (Ore.) Municipal Airport**.

## Design for Production

(Continued from page 20)

performance. How could he be asked to do this? And surely the aircraft operator would not understand the significance of performance in the interests of luxury. "On the air road is the Flying low road in which we will ride or the other in the Devil's Palace which will not fly. What should the compromise be?" Certainly the aircraft must after all conform at least substantially to the basic idea of the long-range monoplane in that connection. Mr. Raymond replied with the conviction that few passengers could partake, and that at any case the other advantages of the low-wing type overtake. In this connection there were Wally Wassman and good words for the use of a shock absorbing wing as a means of getting the load off the tail and having it through the landing. The other replied that as far as shock absorption could be lucrative, the longer slope had the benefit of it, especially as their wings necessarily deflected under load more than those of smaller span.

In the cause of general discussion at the design session, Mr. Raymond announced very favorable results from the testing of a nose wheel on a Douglas amphibian. He suggested that the nose-wheel landing gear might give rise to a much more dominant effect than the tail landing gear. The greater conventional type shock absorber, he said, would be more effective. One operator had complained that the trap surfaces used for the read-

# Aviation People

Who's who and what they are doing

Albert W. Thompson



John W. Shultz



Mark Szwarc



D. W. Remondino



Thor W. Olson



Up the ladder from pilot to executive president goes **Albert W. Thompson**, appointed to that office by National Paris Airways, Inc. He has spent 12 years with the airline, having started as a WPA, prior to air line business in 1931, piloting for National Paris Airways, then its successor, and latterly manager of operations. In his new capacity, Thompson will continue as manager of operations for the line.

• **Trooper** Collector of the Month—**VSA Squadron 102, Commander** **Albert C. McFall**—Highest Serial Memorial Trophy awarded to the Naval Aviation Trooper who has served the flight without stain. **Henry Rosen**, 414B, eighty-two such trophies.

**Doc** **Moskowitz**—**Assistant Air Lines' Distinguished Service Medal** for his record duty trans-Atlantic flight with **Harry Richman**, piloted at a slower speed than the record of **John L. Latimer** and **D. W. Thompson** (assistant to president **Jack Page** at Transoceanic) & **Western Airlines**—U. S. Navy's **Naval Demo Troop**, as behalf of Squadron VN 37 HLD, the Fleet Reserve Aviation Division rated as most efficient for 1936. The winning team was **John T. Tamm** and **Robert A. Price**. **Vivian H. Knecht**—Member of the Legion of Honor, bestowed upon him by the French government when he was in Paris attending the automobile show. **Artie Mac McGinnis**, widow of **Harry Mac McGinnis**, widow of **Harry Mac McGinnis**—**the Distinguished Flying Cross** posthumously for her husband's gallantry in combat. **Mac** had been his partner in the flight of the **Fightless** leaders to Alaska two years ago.

**Captain Frank G. Dowd** and **Captain John S. Gresham**—the Distinguished Flying Cross for heroism during a carbon-monoxide test flight at Wright Field which resulted possible plane fire and destruction of government property. **LaVance R. Remondino**—**Continental Motors Corporation** Trophy for the best design of an airplane in part of the year's design contest at the University of Detroit. **Frank** Remondino's design is a low-powered, two-place aircraft for sport and general use. It has a payload of 100 lbs. **Frank Howes** won second place, **Earle Paauwe**, third. Judges were **Frankie Tamm**, **Chairman**, **Motor Corp.**, **Alfred Dalla Seta**, **Simson Aircraft Corp.**, **M. A. Mills**, **Stout Engineering Laboratories**, **Lt. George G. Price**, **U. S. Naval Reserve**, **Frankie Spragg**, **Hudson Motor Car Co.**

**Thor W. Olson** has been appointed senior manager of **Ex-Cell-O Aircraft & Tool Corporation** of Detroit, succeed- ing **William F. White**. **Thor W. Olson** was general manager, later president, of **Continental Tool Works** until that position was acquired by **Ex-Cell-O**. He became vice-president and a director after the merger.

• Additional honor has been made to the engineering staff of **Kinner Aircraft** &

• An airplane accident over Syracuse, N. Y. on Nov. 15 brought death to **John H. Strode**, Miss **Pauline Mary**, and **Eleanor Strode**. In 1932 Mr. Strode, a prominent architect with P. Arthur Hochschild, president **Brown International**, and set up operations at **Brown Municipal Airport**. Starting with a Stinson, the business expanded and, in 1935, with five planes based in his hangar, it was said Mr. Strode was the nation's first "flying circus." Eighteen aircraft were operated. He was a member of the **N. A. A.** serving as a delegate for Massachusetts in 1932 convention, also an active member of the **Quigley Business** and the **Auto Club of Boston**.

• **Mount** on the air line, checkered: **C. R. H. Weller**, station manager at **Lilac Rock for American Airlines**, became station manager for the line at **Los Angeles**. **Tom McRae** was appointed and dispatched to **Los Angeles** to assume station manager.

• **D. W. Remondino** **American Airlines** first radio operator at **Memphis**, has been made chief operator of the line's system. In his new job he supervises training of radio traffic and operations of radio equipment. **J. J. Nevin** has been appointed manager of **East** **Glendale**, **Calif.**, has been appointed his assistant. **J. C. Tamm** was transferred from **Eastern Air Lines** **Lincolne** station to **Louisville**, Ky., as district traffic manager. **Richard A. Dazzell**, his predecessor, has gone with **Louisville Flying** to **St. Louis** to become manager of passenger division of plane service at **St. Louis**, due to company construction, succeeded **Eastern Air Lines** **transportation** manager **John W. Dacott**. **J. H. Smithson**, in **Charleston** and **Marion** respectively, **W. D. Bessette**, local manager, remains at **St. Louis** to handle the **St. Louis** to **Charleston** passenger transportation of the service. **E. H. Kornblau** has been appointed **Trans-continental & Western Air's** District traffic agent for **New York City**. He was formerly traffic representative. **E. L. Antoniou**, for eight years traffic representative of **Pan American** for **North and South America**, has been made an assistant dispatcher, with headquarters at **Bogota**. His former job at **Pan American** has been temporarily assigned to **Walter Wilson**.

• Additional honor has been made to the engineering staff of **Kinner Aircraft** &



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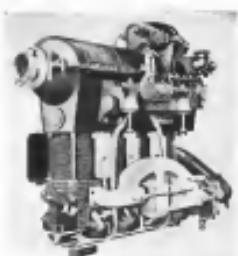
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